

**Fatigue Behavior, Monotonic Properties  
and  
Microstructure Data  
for  
4320, Quenched (Core) Steel  
(Iteration No. 49)**

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March 2002

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## SUMMARY

The required chemical analysis, microstructure data, mechanical properties, cyclic stress-strain data and strain-controlled fatigue data for 4320 Quenched (Core) steel (Iteration # 49) have been obtained. The material was provided by the American Iron and Steel Institute (AISI) in the form of 2.5" bars. These bars were machined into smooth axial fatigue specimens. The specimens were heat treated by the AISI group to reach a hardness of about BHN 188. Two monotonic tensile tests were performed to measure the yield strength, the tensile strength and the reduction of area. Eighteen specimens were fatigue tested in laboratory air at room temperature to establish a strain-life curve.

## INTRODUCTION

This report presents the results of tensile and fatigue tests performed on a group of 20, 4320 Quenched (Core) steel samples. The material was provided by the American Iron and Steel Institute.

The objectives of this investigation were to obtain the chemical analysis, and microstructural data, mechanical properties, cyclic stress-strain data and strain-life fatigue data requested by the AISI bar group.

## EXPERIMENTAL PROCEDURE

### *Specimen Preparation*

The material for the study was received in the form of bars. Smooth cylindrical fatigue specimens, shown in Figure 1, were machined from the metal bars. The gauge sections of the fatigue specimens were mechanically polished in the loading direction using 240, 400, 500, and 600 emery paper. After polishing, a thin band of M-coat D acrylic coating was applied along the central gauge section. The purpose of the M-coat D application was to prevent scratching of the smooth surface by the knife-edges of the strain extensometer, thus reducing the incidence of knife-edge failures. In total, 18 fatigue data points were generated.

### *Test Equipment and Procedure*

Two monotonic tension tests were performed to determine the yield strength, the tensile strength, the percent of elongation and the percent reduction of area. Hardness tests were performed on the surface of three fatigue specimens using a Rockwell B scale. The hardness measurements were repeated three times for each specimen and the average value was recorded.

All fatigue tests were carried out in a laboratory environment at approximately 25 °C using an MTS servo-controlled closed loop electrohydraulic testing machine. A

process control computer, controlled by FLEX software [1] was used to output constant strain and stress amplitudes in the form of a sinusoidal wave.

Axial, constant amplitude, fully reversed ( $R=-1$ ) strain-controlled fatigue tests were performed on smooth specimens. The stress-strain limits for a given cycle of each specimen were recorded at logarithmic intervals throughout the test via a peak reading oscilloscope. Failure of a specimen was defined as a 50 percent drop in tensile peak load from the peak load observed at one half the expected specimen life. For fatigue lives greater than 100,000 reversals, the specimens were tested in stress-control once the stress-strain loops had stabilized. For the stress-controlled tests, failure was defined as the separation of the smooth specimen into two pieces. For strain-controlled tests the loading frequency varied from 0.03 Hz to 5 Hz while in stress-controlled tests the frequency used was up to 110 Hz.

The first reversal of each fatigue test was recorded on a x-y plotter, allowing the elastic modulus (E) and the monotonic yield strength to be determined.

## RESULTS

### A) Microstructure Data

Figure 2 presents the martensite microstructure of the 4320 Quenched (Core) steel. A Type D series inclusion severity level of 1 was obtained based on ASTM E45 (Method A). Inclusions of types A, B and C were not observed. Figure 3 shows the inclusions observed in the 4320 Quenched (Core) steel. The inclusion area was measured using a JAVA image analysis system. The chemical composition of 4320 Quenched (Core) steel was provided by the MacSteel Company, and is shown in Table 1.

### B) Strain-Life Data

The fatigue test data for 4320 Quenched (Core) steel obtained in this investigation are given in Table 2. The stress amplitude corresponding to each strain-amplitude was calculated from the peak load amplitude at the specimen half-life.

A fatigue strain-life curve for the 4320 Quenched (Core) steel is shown in Figure 4, and is described by the following equation:

$$\frac{\Delta\varepsilon}{2} = \frac{\sigma'_f}{E}(2N_f)^b + \varepsilon'_f(2N_f)^c$$

where

- $\frac{\Delta\varepsilon}{2}$  = True total strain amplitude
- $2N_f$  = Number of reversals to failure
- $\sigma'_f$  = Fatigue strength coefficient
- $b$  = Fatigue strength exponent
- $\varepsilon'_f$  = Fatigue ductility coefficient
- $c$  = Fatigue ductility exponent

Where  $\sigma'_f = 909$  MPa,  $b = -0.0355$ ,  $\varepsilon'_f = 0.86$  and  $c = -0.65$ . These values of the strain-life parameters were determined from fatigue testing over the range:  $0.0025 < \frac{\Delta\varepsilon}{2} < 0.010$ .

### C) Cyclic Stress-Strain Curves

Stabilized and half-life stress data obtained from strain-life fatigue tests were used to obtain the companion cyclic stress-strain curve shown in Figure 5. The cyclic stress-strain curve is described by the following equation:

$$\varepsilon = \frac{\sigma}{E} + \left(\frac{\sigma}{K'}\right)^{\frac{1}{n'}}$$

where

- $\varepsilon$  = True total strain amplitude
- $\sigma$  = Cyclically stable true stress amplitude
- $K'$  = Cyclic strength coefficient
- $n'$  = Cyclic strain hardening exponent

Where  $K' = 799$  MPa and  $n' = 0.032$ .

## **D) Mechanical Properties**

The engineering monotonic stress-strain curve is given in Figure 6. The monotonic and cyclic properties are included in Appendix 1. The Hardness of the 4320 Quenched (Core) steel was taken as the average of three randomly chosen fatigue specimens and is given in Appendix 1. The individual hardness measurements are also given in Table 2. The true monotonic and true cyclic stress-strain curves plotted together are given in Figure 7.

## **REFERENCES**

- [1] Pompetzki, M.A., Saper, R.A., and Topper, T.H., "Software for High Frequency Control of Variable Amplitude Fatigue Tests," *Canadian Metallurgical Quarterly*, Vol. 25, No. 2, pp. 181-194, 198.
- [2] J. A. Bannantine, J. J. Comer, and J. L. Handrock (1990), In :*Fundamentals of Metal Fatigue Analysis*, Prentice Hall, London.

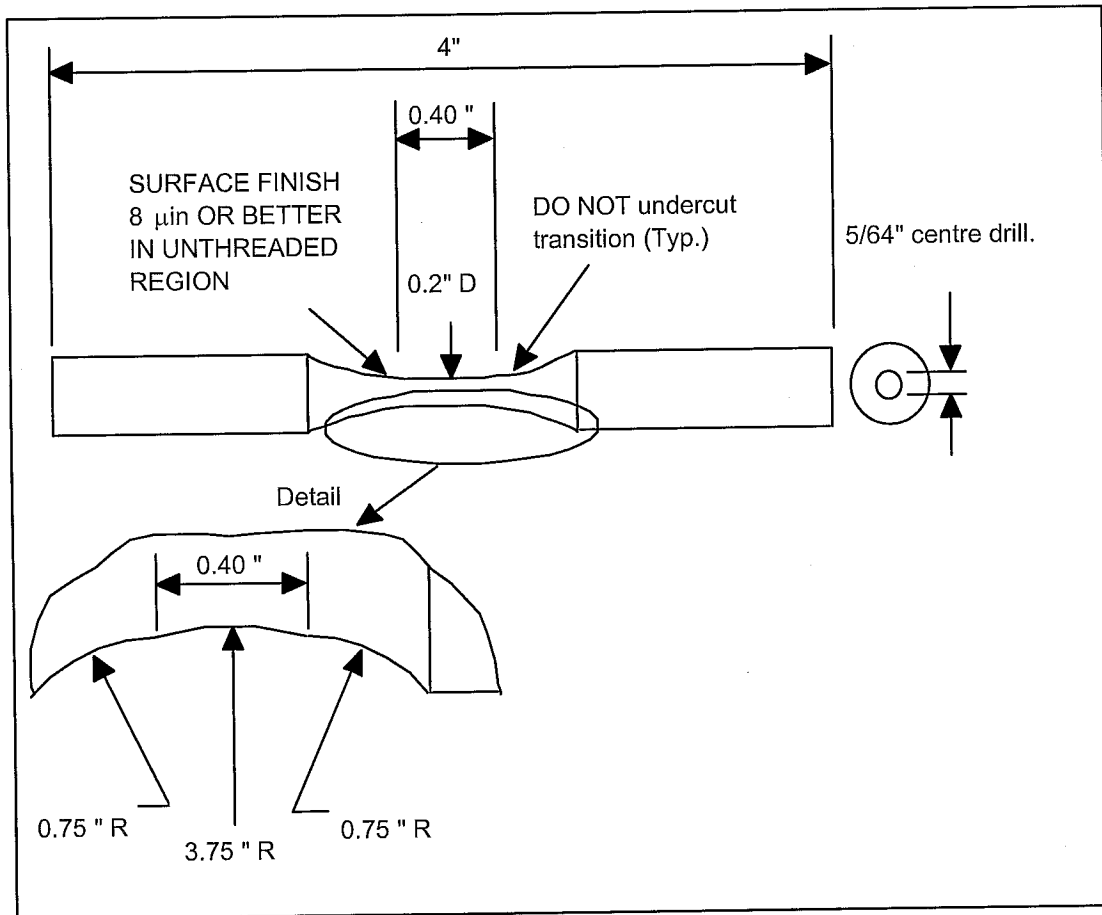
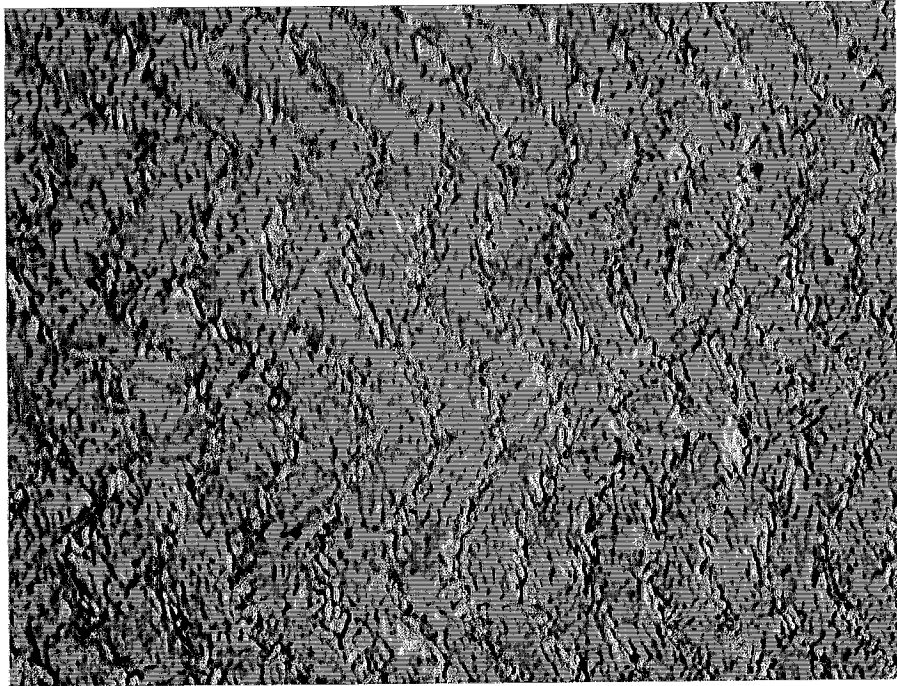
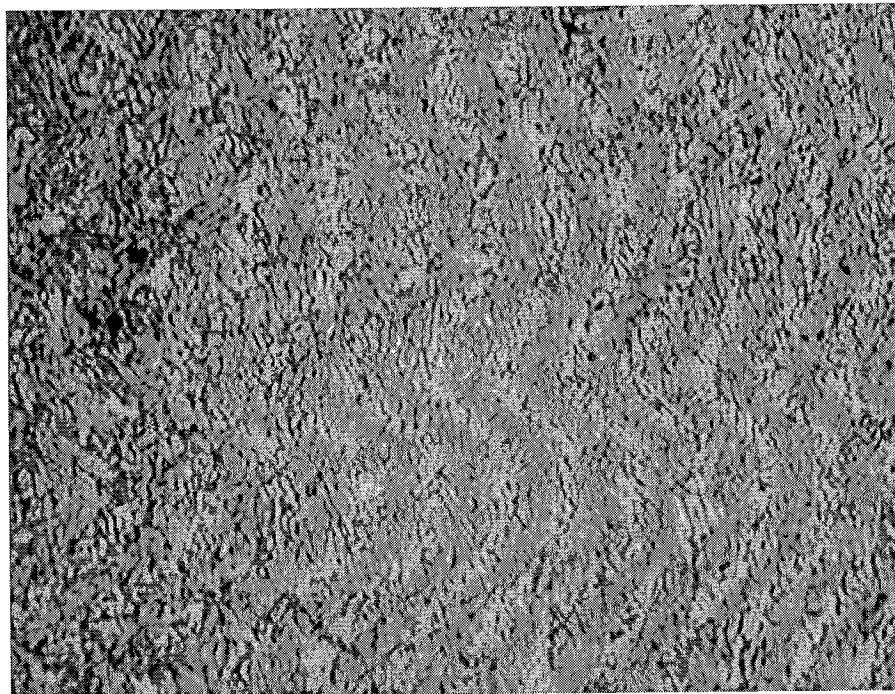


Fig. 1 Smooth cylindrical fatigue specimen





(a) Longitudinal Direction



(b) Transverse Direction

Figure 2. Photomicrographs of 4320 Quenched (Core) steel (X500)

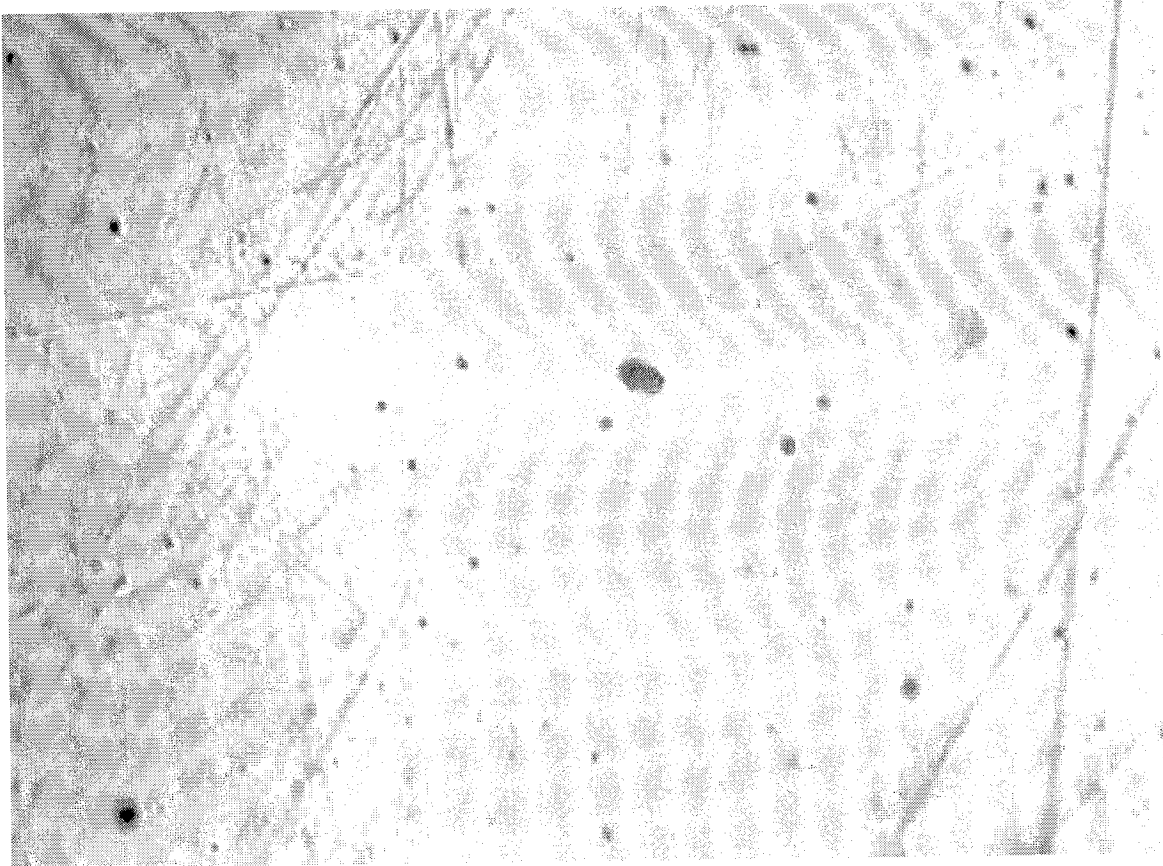


Figure 3. Inclusions photomicrograph of 4320 Quenched (Core) steel (X100)

### 4320 Quenched (Core) Steel

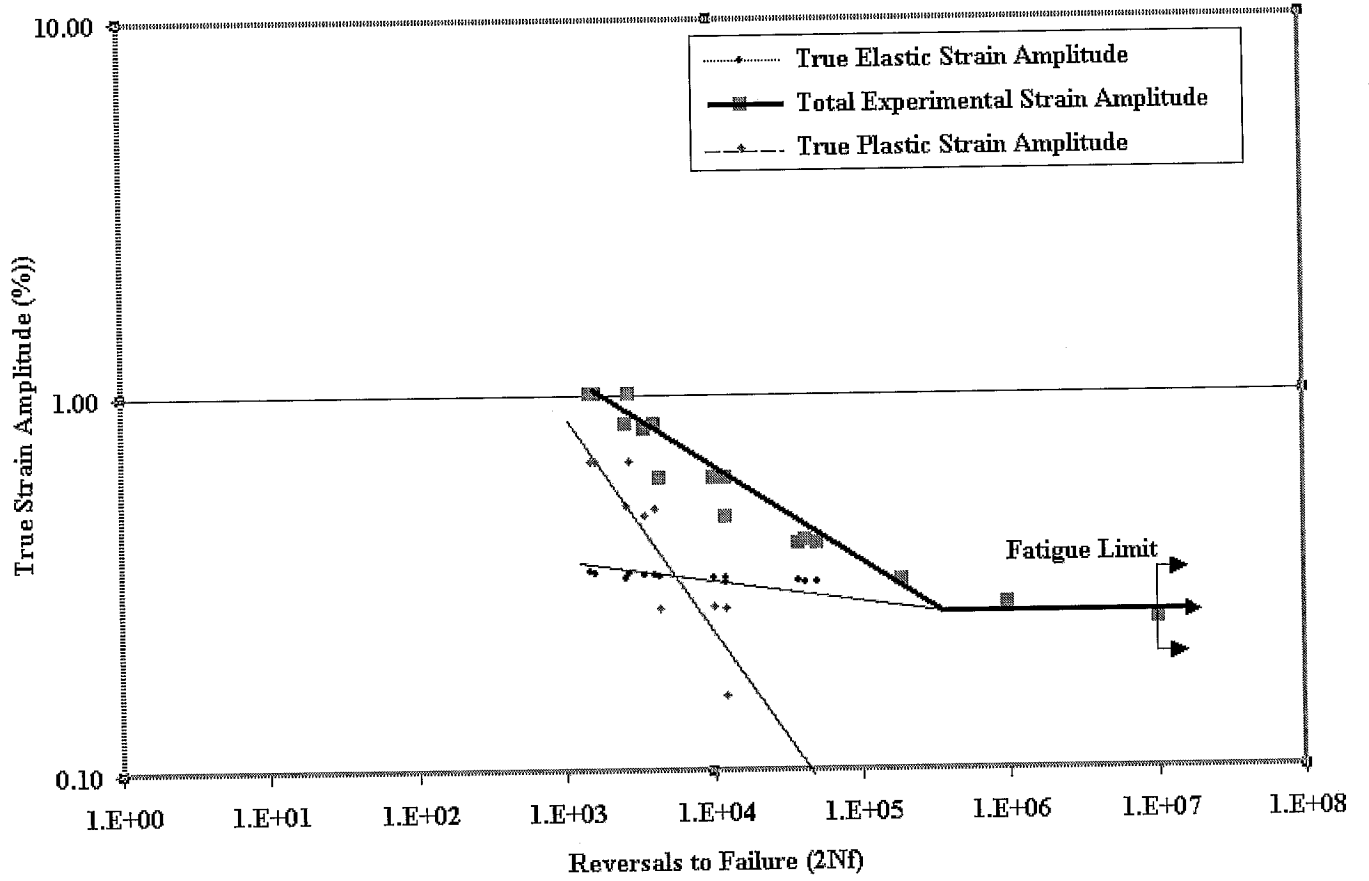


Figure 4. Constant amplitude fully reversed strain-life curve for 4320 Quenched (Core) steel.

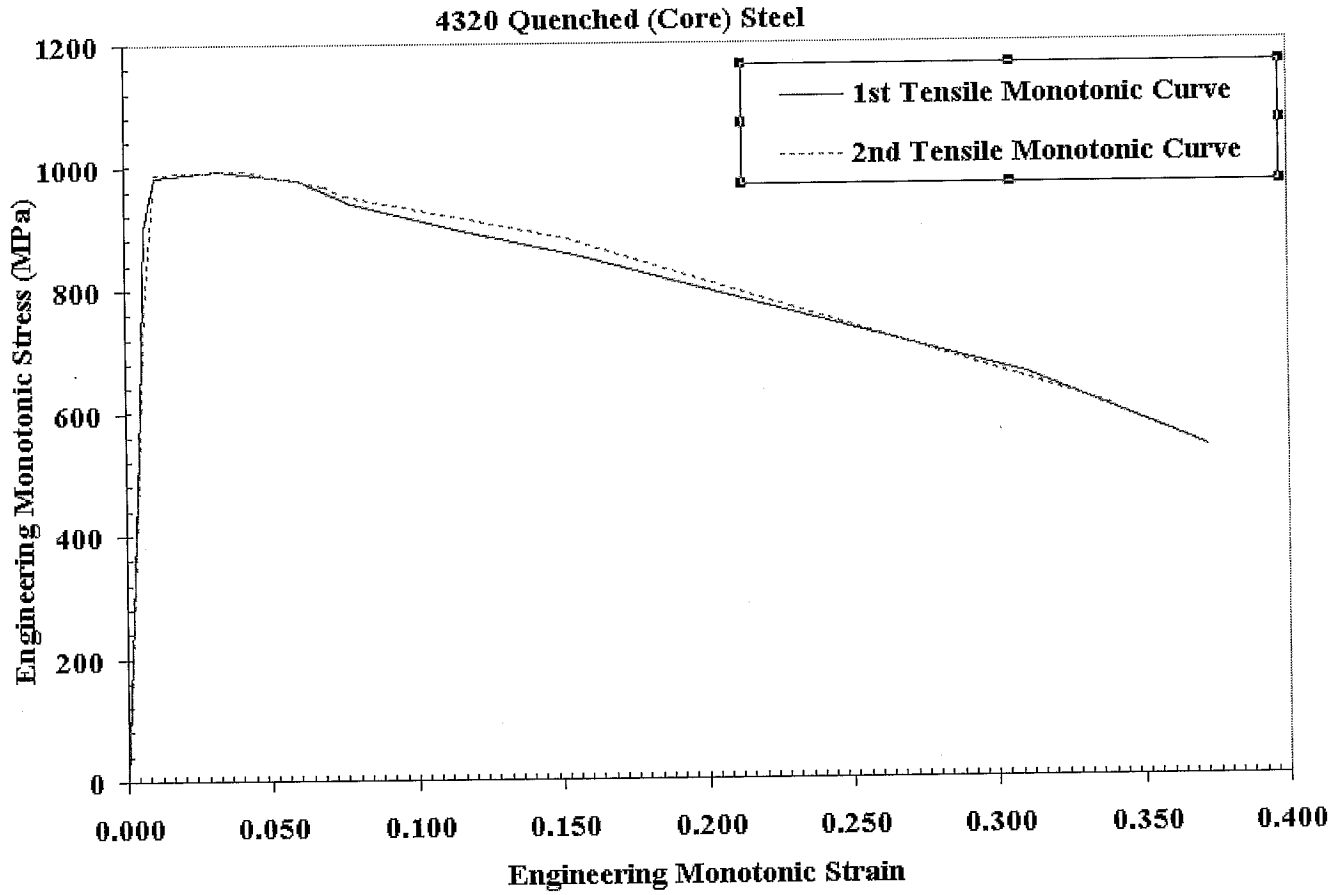


Figure 5. Monotonic stress-strain curves for two 4320 Quenched (Core) steel specimens.

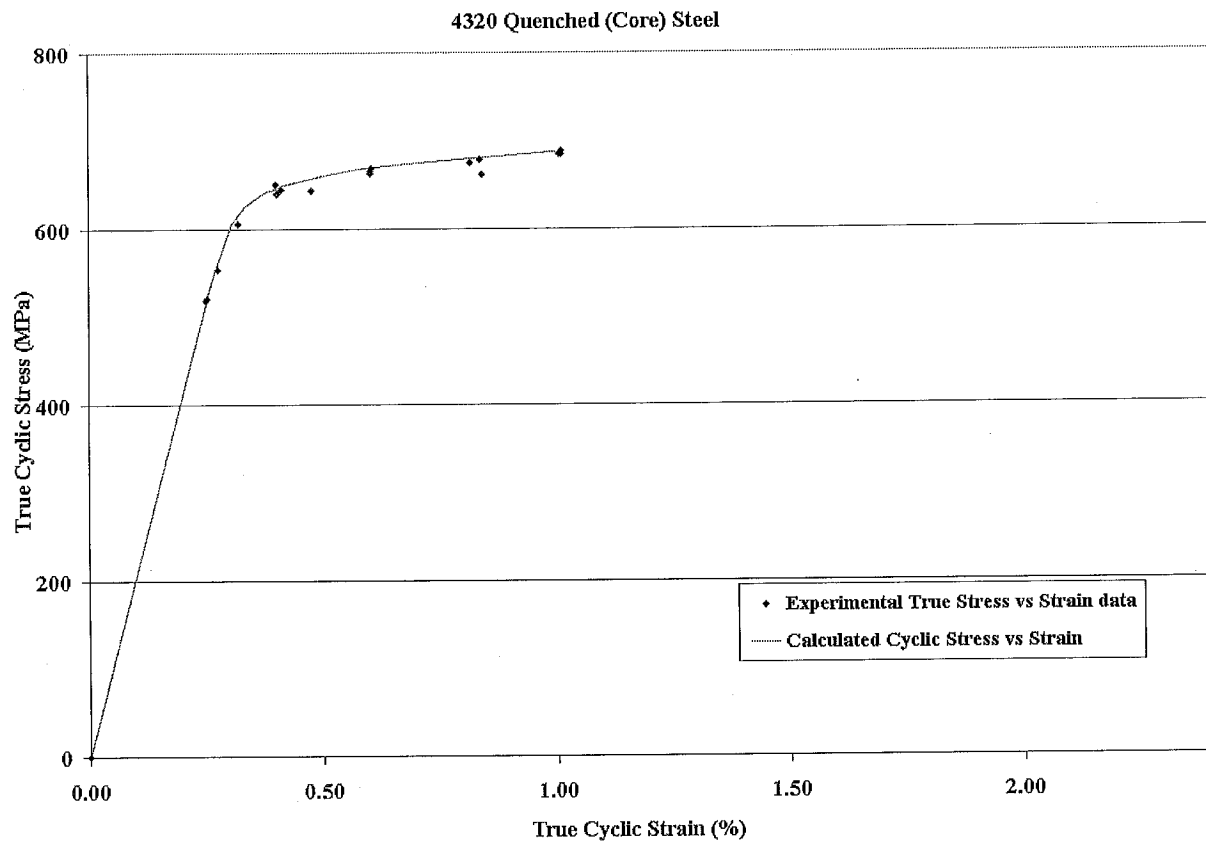


Figure 6. Cyclic stress-strain curve for 4320 Quenched (Core) steel.

4320 Quenched (Core) Steel

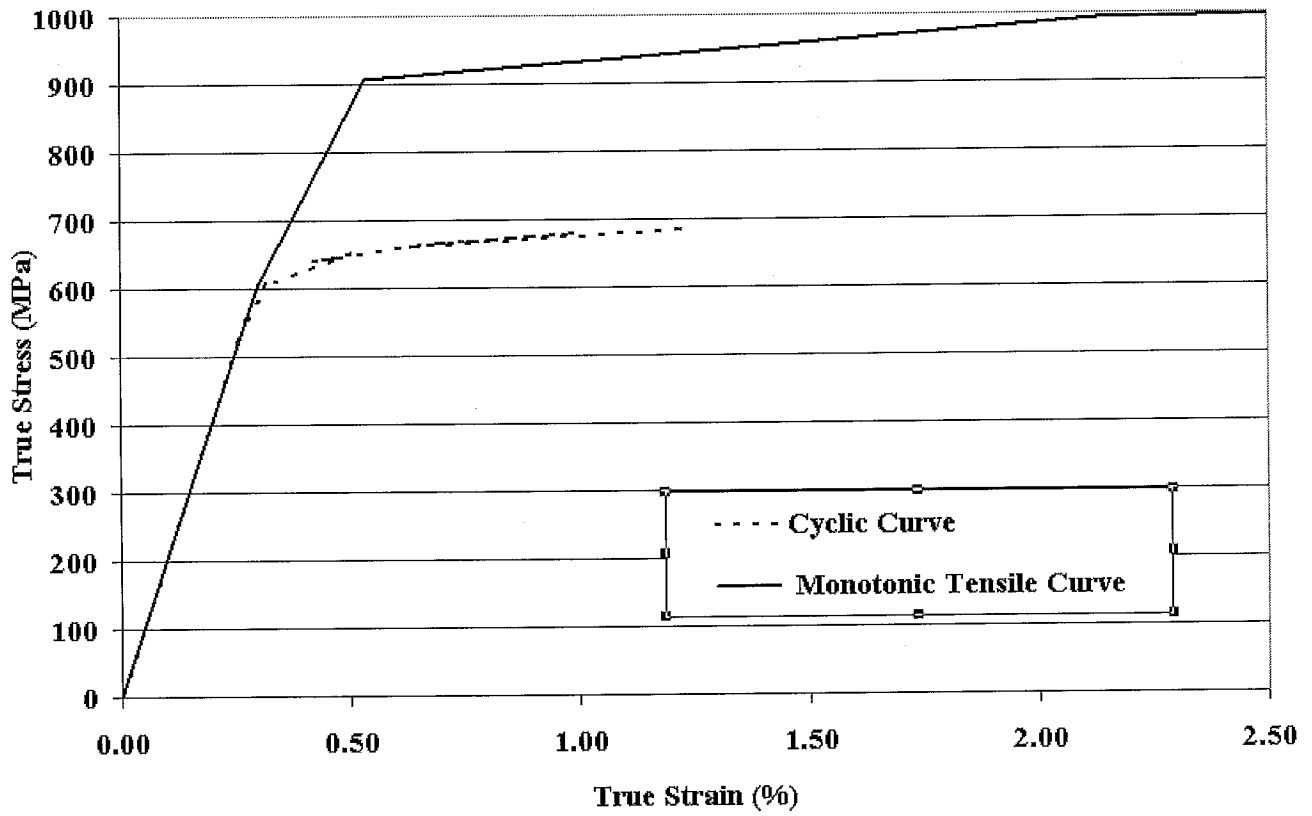


Fig. 7 Monotonic and Cyclic stress-strain curves for 4320 Quenched (Core) steel.

**Table 1 Chemical composition of 4320 Quenched (Core) steel.**

Carbon, C	0.18%
Manganese, Mn	0.48%
Phosphorous, P	0.006%
Sulfur, S	0.02%
Silicon, Si	0.26%
Copper, Cu	0.17%
Nickel, Ni	1.74%
Chromium, Cr	0.58%
Molybdenum, Mo	0.17%
Sn	NA
Al	NA
Vanadium, Va	NA
N	NA
Ti	NA
N2	NA
V	0.002%
Cb	NA
Te	NA

Table 2 Tensile and Fatigue Test Data for 4320 Quenched (Core) steel.

Sp#	Total Strain Amplitude(%)	Stress Amplitude (MPa)	Plastic Strain Amplitude(%)	Elastic Strain Amplitude(%)	(50% load drop) Fatigue Life (Reversals, 2Nf)	MONOTONIC Young's Modulus(GPa)	Hardness (HRB)
2	1.009	688	0.668	0.341	1480	204	92
4	1.009	685	0.670	0.339	2742	201	92
3	1.006	685	0.667	0.339	1600	203	94
7'	0.839	662	0.512	0.327	2580	203	94
8	0.836	678	0.501	0.336	4014	191	99
14'	0.816	675	0.482	0.334	3436	201	99
13	0.604	668	0.273	0.331	4358	204	92
12	0.602	666	0.273	0.329	12240	201	96
15	0.603	663	0.275	0.328	10140	203	92
5'	0.477	643	0.159	0.318	12258	201	93
10	0.413	645	0.094	0.319	41986	203	92
1	0.405	639	0.088	0.317	50060	204	95
11	0.401	650	0.079	0.322	38216	202	90
6	0.320	606	0.000	0.320	188026	201	99
9	0.277	553	0.000	0.277	979058	202	99
5*	0.253	520	0.000	0.253	10000000	201	92
7*	0.252	517	0.000	0.252	10000000	203	96
14*	0.252	517	0.000	0.252	10000000	201	94

\* Run out



## Appendix 1

### Monotonic Properties for 4320 Quenched (Core) steel.

Average Elastic Modulus, E	=	202	GPa
Yield Strength	=	920	MPa
Ultimate tensile Strength	=	994	MPa
% Elongation	=	43	%
% Reduction of Area	=	63	%
True fracture strain, $Ln (A_i / A_f)$	=	99	%
True fracture stress, $\sigma_f = \frac{P_f}{A_f}$	=	1520	MPa
Bridgman correction, $\sigma_f = \frac{P_f}{A_f} \left/ \left( 1 + \frac{4R}{D_f} \right) \right. Ln \left( 1 + \frac{D_f}{4R} \right)$			= 1336 MPa
Monotonic strength coefficient, K	=	1127	MPa
Monotonic strain hardening exponent, n	=	0.0308	
Hardness, Rockwell B (HRB)	=	94	
Hardness, Brinell	=	188	

### Cyclic Properties for 4320 Quenched (Core) steel.

Cyclic Yield Strength, (0.2% offset) = $K'(0.002)^{n'}$	=	652	MPa
Cyclic strength coefficient, K'	=	799	MPa
Cyclic strain hardening exponent, n'	=	0.032	
Fatigue Strength Coefficient, $\sigma'_f$	=	909	MPa
Fatigue Strength Exponent, b	=	-0.035	
Fatigue Ductility Coefficient, $\epsilon'_f$	=	0.86	
Fatigue Ductility Exponent, c	=	-0.65	

P <sub>f</sub> :	Load at fracture.
A <sub>i</sub> and A <sub>f</sub> :	Specimen cross-section area before and after fracture.
R:	Specimen neck radius.
D <sub>f</sub> :	Specimen diameter at fracture.