Fatigue Behavior, Monotonic Properties and

Microstructure Data

for

SAE 9254 AL FG, Quenched and Tempered

(Iteration No. 35)

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SUMMARY

The required chemical analysis, microstructure data, mechanical properties, cyclic stress-strain data and strain-controlled fatigue data for SAE 9254 AL FG, Quenched and Tempered (Iteration No. 35) have been obtained. The material was provided by the American Iron and Steel Institute (AISI) in the form of metal bars. These bars were machined into smooth axial fatigue specimens. The specimens were heat treated to temperature 930 C° in nitrogen media and quenched in oil and tempered to give a hardness of about Rc 58. Two monotonic tensile test was performed to measure yield strength, tensile strength and reduction of area. Eighteen specimens were fatigue tested in laboratory air at room temperature to establish a strain-life curve.

INTRODUCTION

This report presents the results of tensile and fatigue tests performed on a group of 19 SAE 9254 AL FG, Quenched and Tempered steel samples. The material was provided by the American Iron and Steel Institute.

The objectives of this investigation were to obtain the chemical analysis, and microstructural data, mechanical properties, cyclic stress-strain data and strain-life tests requested by the AISI bar group.

EXPERIMENTAL PROCEDURE

Specimen Preparation

The material for the study was received in the form of bars. Smooth cylindrical fatigue specimens, shown in Figure 1, were machined from the metal bars then sent to be heat treated at Cambridge Heat Treatment Inc. The gauge sections of the fatigue specimens were mechanically polished in the loading direction using 240, 400, 500, and 600 emery paper. After polishing, a thin band of M-coat D acrylic coating was applied along the central gauge section. The purpose of the M-coat D application was to prevent scratching of the smooth surface by the knife-edges of the strain extensometer, thus reducing the incidence of knife-edge failures. In total, 18 fatigue data points were generated.

Test Equipment and Procedure

Two monotonic tension tests were performed to determine the yield strength, the tensile strength, the percent of elongation and the percent reduction of area. Hardness tests were performed on the surface of three fatigue specimens using a Rockwell C scale. The hardness measurements were repeated three times for each specimen and the average value was recorded.

All fatigue tests were carried out in a laboratory environment at approximately 25 °C using an MTS servo-controlled closed loop electrohydraulic testing machine. A

process control computer, controlled by FLEX software [1] was used to output constant strain and stress amplitudes in the form of a sinusoidal wave.

Axial, constant amplitude, fully reversed (R=-1) strain-controlled fatigue tests were performed on smooth specimens. The stress-strain limits for a given cycle of each specimen were recorded at logarithmic intervals throughout the test via a peak reading oscilloscope. Failure of a specimen was defined as a 50 percent drop in tensile peak load from the peak load observed at one half the expected specimen life. For fatigue lives greater than 100,000 reversals, the specimens were tested in stress-control once the stress-strain loops had stabilized. For the stress-controlled tests, failure was defined as the separation of the smooth specimen into two pieces. For strain-controlled tests the loading frequency varied from 0.03 Hz to 5 Hz while in stress-controlled tests the frequency used was up to 80 Hz.

The first reversal of each fatigue test was recorded on a x-y plotter, allowing the elastic modulus (E) and the monotonic yield strength to be determined.

RESULTS

A) Microstructure Data

Figure 2 presents the martensitic microstructure of SAE 9254 AL FG, Quenched and Tempered steel. A Type A and D inclusion severity level of $1\frac{1}{2}$ was obtained based on ASTM E45 (Method A). It was observed that 78% of type D inclusions were of the thin series and 22% were of the thick series. Inclusions of types B, and C were not observed. Figure 3 shows the inclusions observed in the SAE 9254 AL FG, Quenched and Tempered steel. The inclusion area was measured using a JAVA image analysis system. The chemical composition of SAE 9254 AL FG, Quenched and Tempered steel was provided by the supplier (Stelco Inc.), and is shown in Table 1.

B) Strain-Life Data

The fatigue test data for SAE 9254 AL FG, Quenched and Tempered steel obtained in this investigation are given in Table 2. The stress amplitude corresponding to each strain-amplitude was calculated from the peak load amplitude at the specimen half-life.

A fatigue strain-life curve for the SAE 9254 AL FG, Quenched and Tempered steel is shown in Figure 4, and is described by the following equation:

$$\frac{\Delta \varepsilon}{2} = \frac{\sigma_f'}{E} (2N_f)^b + \varepsilon_f' (2N_f)^c$$

where

 $\frac{\Delta \varepsilon}{2}$ = True total strain amplitude

 $2N_f$ = Number of reversals to failure

 σ'_{f} = Fatigue strength coefficient

b = Fatigue strength exponent

 ε' = Fatigue ductility coefficient

c = Fatigue ductility exponent

Where $\sigma'_f = 4108$ MPa, b = -0.109, $\varepsilon'_f = 1.13$ and c = -0.954. These values of the strain-life parameters were determined from fatigue testing over the range: $0.0032 < \frac{\Delta \varepsilon}{2} < 0.01$.

C) Cyclic Stress-Strain Curves

Stabilized and half-life stress data obtained from strain-life fatigue tests were used to obtain the companion cyclic stress-strain curve shown in Figure 5. The true cyclic stress-strain curve is described by the following equation:

$$\varepsilon = \frac{\sigma}{E} + \left(\frac{\sigma}{K'}\right)^{\frac{1}{n'}}$$

where

 ε = True total strain amplitude

 σ = Cyclically stable true stress amplitude

K' = Cyclic strength coefficient
 n' = Cyclic strain hardening exponent

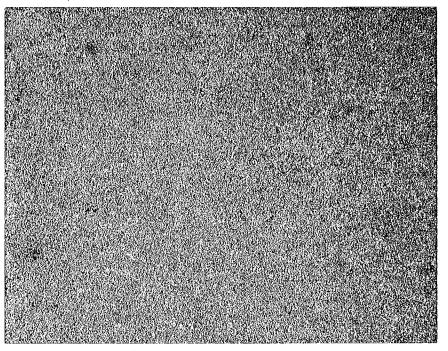
Where K' = 3322 MPa and n' = 0.088.

D) Mechanical Properties

The engineering monotonic stress-strain curve is given in Figure 6. The monotonic and cyclic properties are included in Appendix 1. The Hardness of the SAE 9254 AL FG, Quenched and Tempered steel taken as the average of three randomly chosen fatigue specimens and is given in Appendix 1. The individual hardness measurements are also given in Table 2. The true monotonic and true cyclic stress-strain curves plotted together are given in Figure 7.

REFERENCES

- [1] Pompetzki, M.A., Saper, R.A., and Topper, T.H., "Software for High Frequency Control of Variable Amplitude Fatigue Tests," Canadian Metallurgical Quarterly, Vol. 25, No. 2, pp. 181-194, 198.
- [2] J. A. Bannantine, J. J. Comer, and J. L. Handrock (1990), In: Fundamentals of Metal Fatigue Analysis, Prentice Hall, London.



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Fig. 1-2 Iter. 35, 9254 Steel, Quenched and tempered to Rc 53 Upper- 100X Lower- 500X Nital Etched

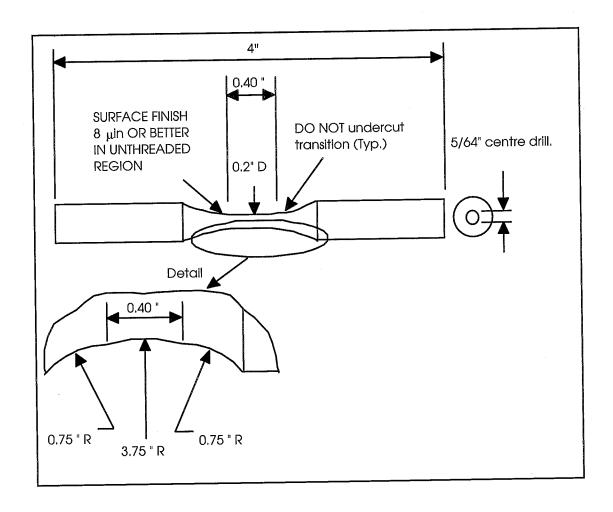
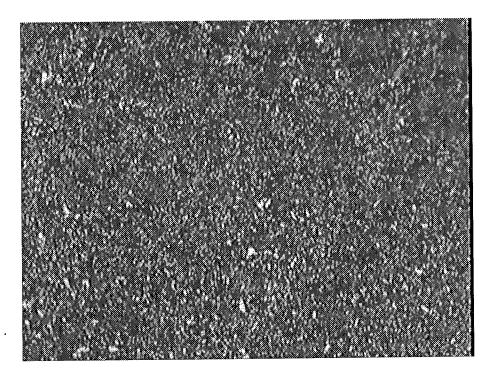
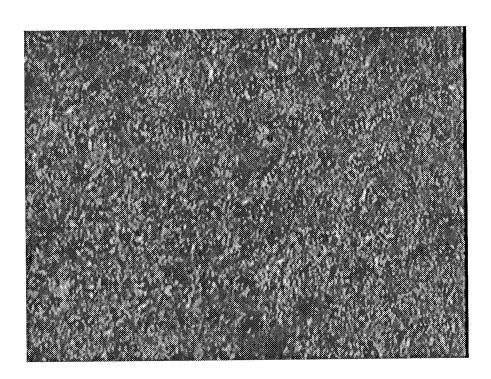


Fig. 1 Smooth cylindrical fatigue specimen



(a) Longitudinal Direction



(b) Transverse Direction

Fig. 2 Photomicrographs of SAE 9254 AL FG, Quenched and Tempered (X200)

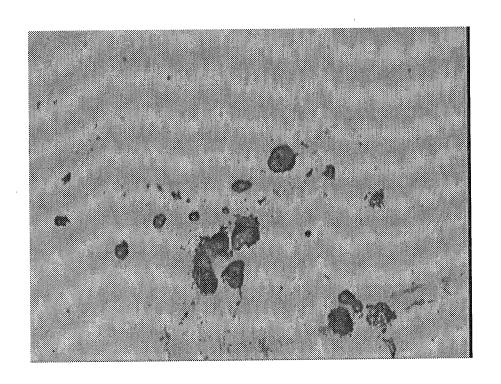


Fig. 3 Inclusions photomicrograph of SAE 9254 AL FG, Quenched and Tempered steel (X500)

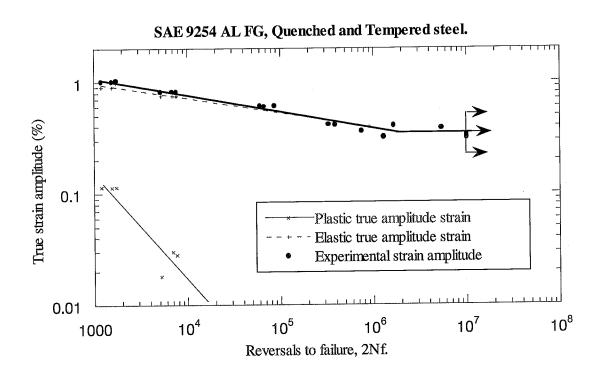


Fig. 4 Constant amplitude fully reversed strain-life curve for SAE 9254 AL FG, Quenched and Tempered steel.

SAE 9254 Al FG, Quenched & Tempered

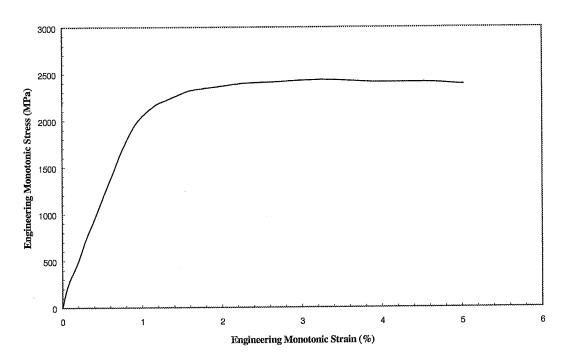


Fig. 5 Monotonic stress-strain curve for SAE 9254 AL FG, Quenched and Tempered steel.

SAE 9254 Al FG, Quenched & Tempered

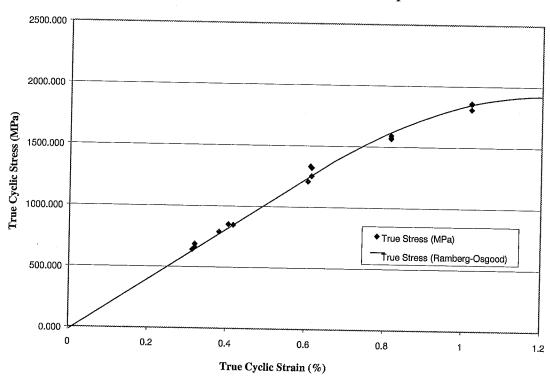


Fig. 6 Cyclic stress-strain curve for SAE 9254 AL FG, Quenched and Tempered steel.

SAE 9254 AL FG, Quenched and Tempered steel

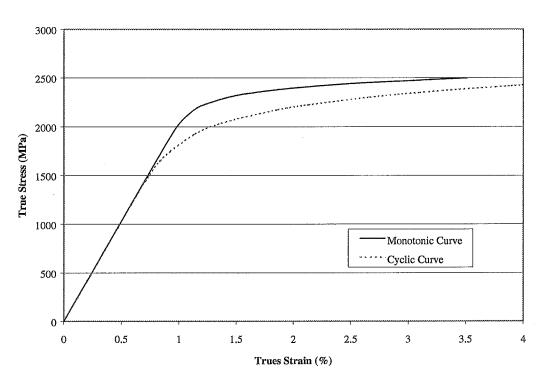


Fig. 7 Monotonic and Cyclic stress-strain curves for SAE 9254 AL FG, Quenched and Tempered steel.

Table 1 Chemical composition of SAE 9254 AL FG, Quenched and Tempered steel.

Carbon, C	0.57%
Manganese, Mn	0.71%
Phosphorous, P	0.011%
Sulfur, S	0.009%
Silicon, Si	1.57%
Copper, Cu	0.011%
Nickel, Ni	0.01%
Chromium, Cr	0.72%
Molybdenum, Mo	0.005%
Vanadium, Va	0.007%
ASA	0.028%
N	0.007%
Columbium, Cb	0.002%

Table 2 Tensile and Fatigue Test Data for SAE 9254 AL FG, Quenched and Tempered steel.

Hardness	(HRC)	29	58	57	58	58	28	2 9	8 %	000	000	28	28	57	57	58	58	7	72	28
MONOTONIC Young's	Modulus(GFa)	206.8	203	205	201	207	206	212	202	202	2007	200.0	700.1	200	202	207	208	206	203	206.3
(50% load drop) Fatigue Life (Peysers) 2NF	(INCVCISAIS, ZIVI)	07/1	1550	1200	6926	7628	5202	60194	87254	0.02/29	370726	164500	104322	392004	5432572	745818	10000000	1284974	10000000	10000000
Elastic Strain Amplitude(%)	0.901	0.000	0.907	0.901	0.786	0.788	0.798	0.615	0.612	909.0	0.417	0.405	0.40	0.410	0.381	0.360	0.330	0.320	0.320	0.314
Plastic Strain Amplitude(%)	0.114	0.113	0.113	411.0	0.030	0.028	0.018	0.000	0.000	0.000	0.000	0000	0000	0000	0.000	0.000	0.000	0.000	0.000	0.000
Stress Amplitude (MPa)	1854	1860	1809	1565	1500	0/51	1590	1254	1333	1208	845	851	840	780	109	/14	1 /89	L 89	661 -	642 <-
Total Strain Amplitude(%)	1.015	1.015	1.015	0.813	0.013	0.010	0.810	0.615	0.612	909.0	0.417	0.405	0.410	0.381	0.360	0.300	0.330	0.320	0.320	0.314
#dS	11	12	19	<u>×</u>	10	2 0	۱, ۷	/1	∞	15	_	7	70	'n	, <	† *	<u>+</u> 1	n į	13*	*9

* Run out

Cations Stra

Appendix 1

Monotonic Properties for SAE 9254 AL FG, Quenched and Tempered steel.

Average Elastic Modulus, E	=	205.4	GPa
Yield Strength	=	2270	MPa
Ultimate tensile Strength			
% Elongation	=	2950	MPa
	=	3.92	%
% Reduction of Area	=	3.99	%
True fracture strain, $Ln (A_i/A_f)$		4.08	%
True fracture stress, $\sigma_f = \frac{P_f}{A_f}$	= '	2483	MPa
Bridgman correction, $\sigma_f = \frac{P_f}{A_f} / \left(1 + \frac{4R}{D_f}\right)$	$-\int Ln\bigg(1+$	$\left(\frac{D_f}{4R}\right) = 20$)47MPa
Monotonic strength coefficient, K	´ =		MPa
Monotonic strain hardening exponent, n	=	0.0418	1411 a
Hardness, Rockwell C (HRC)			
	=	58	
Hardness, Brinell	=	584	

Cyclic Properties for SAE 9254 AL FG, Quenched and Tempered steel.

Cyclic Yield Strength, $(0.2\% \text{ offset}) = K'(0.2\% \text{ offset})$	$(002)^{n'}$	= 1922	MPa
Cyclic strength coefficient, K'	=	3322	MPa
Cyclic strain hardening exponent, n'	=	0.088	IVII. U
Fatigue Strength Coefficient, σ'	=	4108	MPa
Fatigue Strength Exponent, b	=	-0.109	1111 4
Fatigue Ductility Coefficient, ε'	=	1.13	
Fatigue Ductility Exponent, c	-	-0.954	
2 - Ponent, C	_	-0.934	

 P_r : A_i and A_r : R: Load at fracture.

Specimen cross-section area before and after fracture.

Specimen neck radius.

 D_{f} Specimen diameter at fracture.