AISI 1090M Hot Formed-Accelerated Cooled Steel Iteration #10

Fatigue Behavior, Monotonic Properties and Microstructural Data

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SUMMARY

The required chemical analysis, microstructural data, mechanical properties, cyclic stress-strain data and strain-controlled fatigue data for AISI 1090M-Accelerated Cooled steel (Iteration # 9) have been obtained. The material was provided by the American Iron and Steel Institute (AISI) in the form of metal bars. These bars were machined into smooth axial fatigue specimens. A monotonic tensile test was performed to measure yield strength, tensile strength and reduction of area. Twenty specimens were fatigue tested in laboratory air at room temperature to establish a strain-life curve.

INTRODUCTION

This report presents the results of tensile and fatigue tests performed on a group of 20 1090M-Accelerated Cooled steel samples. The material was provided by the American Iron and Steel Institute.

The objectives of this investigation were to obtain a chemical analysis, and the microstructural data, mechanical properties, cyclic stress-strain data and strain-life tests requested by the AISI bar group.

EXPERIMENTAL PROCEDURE

Specimen Preparation

The material for the study was received in the form of bars. Smooth cylindrical fatigue specimens, shown in Figure 1, were machined from the metal bars. The gauge sections of the fatigue specimens were mechanically polished in the loading direction using 240, 400, 500, and 600 emery paper. After polishing, a thin band of M-coat D acrylic coating was applied along the central gauge section. The purpose of the M-coat D application was to prevent scratching of the smooth surface by the knife edges of the strain extensometer, thus reducing the incidence of knife-edge failures. In total, 19 fatigue data points were generated.

Test Equipment and Procedure

A monotonic tension test was performed to determine the yield strength, the tensile strength, the percent of elongation and the percent reduction of area. Hardness tests were performed on the surface of three fatigue specimens using a Rockwell C scale. The hardness measurements were repeated three times for each specimen and the average value was recorded.

All fatigue tests were carried out in a laboratory environment at approximately 25 °C using an MTS servo-controlled closed loop electrohydraulic testing machine. A process control computer, controlled by FLEX software [1] was used to output constant strain and stress amplitudes in the form of a sinusoidal wave.

Axial, constant amplitude, fully reversed (R=-1) strain-controlled fatigue tests were performed on smooth specimens. The stress-strain limits for a given cycle of each specimen were recorded at logarithmic intervals throughout the test via a peak reading oscilloscope. Failure of a specimen was defined as a 50 percent drop in tensile peak load from the peak load observed at one half the expected specimen life. For fatigue lives greater than 100,000 reversals, the specimens were tested in stress-control once the stress-strain loops had stabilized. For the stress-controlled tests, failure was defined as the separation of the smooth specimen into two pieces. For strain-controlled tests the loading frequency varied from 0.03 Hz to 5 Hz while in stress-controlled tests the frequency used was up to 30 Hz.

The first reversal of each fatigue test was recorded on an x-y plotter, allowing the elastic modulus (E) and the monotonic yield strength to be determined.

RESULTS

A) Microstructural Data

Figure 2 presents the Martensite/Bainite microstructure of AISI 1090M-Accelerated Cooled steel. A type A inclusion rate of $\frac{1}{2}$ was obtained based on the severity level number according to ASTM E45 method A. Inclusions of types B, C, and D were not observed. Figure 3 presents the observed inclusions of AISI 1090Ms Accelerated Cooled steel. The inclusion area was measured using a JAVA image analysis system. The chemical composition of AISI 1090M-Accelerated Cooled steel was provided by SCI-Lab materials testing inc., 25 McIntyre place, unit 2, Kitchener, Ontario, N2R 1H1, and is shown in Table 1.

B) Strain-Life Data

The fatigue test data for AISI 1090M (GM) Hot Form-Accelerated Cooled steel obtained in this investigation are given in table 2. The stress amplitude corresponding to each strain-amplitude was calculated from the peak load amplitude at one half the expected specimen life.

A fatigue strain-life curve for the AISI 1090M (GM) Hot Form-Accelerated Cooled steel is shown in Figure 4, and may is described by the following equation:

$$\frac{\Delta \varepsilon}{2} = \frac{\sigma_{\rm f}'}{E} (2N_{\rm f})^{\rm b} + \varepsilon_{\rm f}' (2N_{\rm f})^{\rm c}$$

where

 $\frac{\Delta \varepsilon}{2}$ = True total strain amplitude

 $2N_f$ = Number of reversals to failure

 $\sigma'_{\mathbf{f}}$ = Fatigue strength coefficient

b = Fatigue strength exponent

 $\mathbf{E}'_{\mathbf{f}}$ = Fatigue ductility coefficient

c = Fatigue ductility exponent

Where $\sigma'_{\mathbf{f}}$ =1715.16 MPa, \mathbf{b} = -0.1046, $\epsilon'_{\mathbf{f}}$ =0.7978 and \mathbf{c} = -0.5996. These values of the strain-life parameters were determined from fatigue testing over the range: 0.002 < $\frac{\Delta\epsilon}{2}$ < 0.01.

C) Cyclic Stress-Strain Curves

Stabilized and half life stress data obtained from strain-life fatigue tests were used to obtain the companion cyclic stress-strain curve shown in figure 5. The true cyclic stress-strain curve is described by the following equation:

$$\varepsilon = \frac{\sigma}{E} + \left(\frac{\sigma}{K'}\right)^{\frac{1}{n'}}$$

where

 ε = True total strain amplitude

 σ = Cyclically stable true stress amplitude

K' = Cyclic strength coefficient

n' = Cyclic strain hardening exponent

Where K' = 1964.20 MPa and n' = 0.1903.

D) Mechanical Properties

The engineering monotonic stress-strain curve is given in figure 6. The monotonic and cyclic properties are included in Appendix 1. The Hardness of the AISI 1090M (GM) Hot Form-Accelerated Cooled steel taken as the average of three randomly chosen fatigue specimens and is given in Appendix 1. The individual hardness measurements are also given in Table 2. The true monotonic and true cyclic stress-strain curves plotted together are given in figure 7.

REFERENCES

- [1] Pompetzki, M.A., Saper, R.A., and Topper, T.H., "Software for High Frequency Control of Variable Amplitude Fatigue Tests," Canadian Metallurgical Quarterly, Vol. 25, No. 2, pp. 181-194, 198.
- [2] J. A. Bannantine, J. J. Comer, and J. L. Handrock (1990), In: Fundamentals of Metal Fatigue Analysis, Prentice Hall, London.

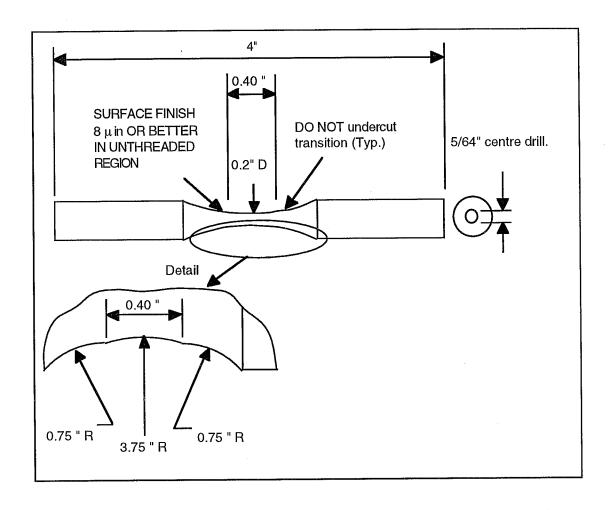
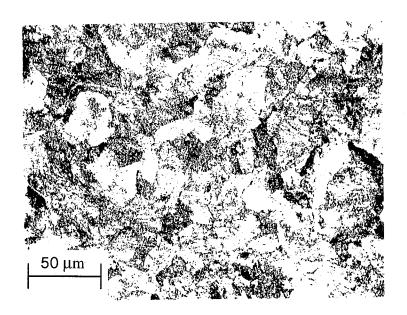
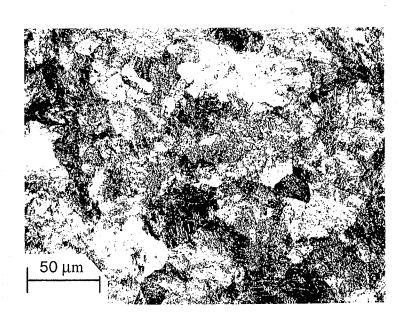


Fig. 1 Smooth cylindrical fatigue specimen



ITER 10: Photomicrograph of SAE 1090 steel, Austempered to Rc-29. 500X Mag.

(a) Longitudinal direction



(b) Transverse direction

Fig. 2 Photomicrographs of AISI 1090 quenched and tempered steel (X500): (a) Longitudinal direction, and (b) Transverse direction.

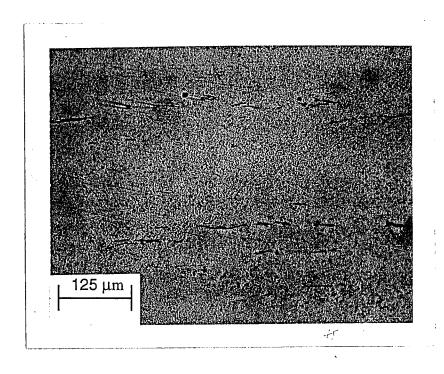


Fig. 3 Inclusions photomicrograph of AISI 1090M-Accelerated Cooled steel (X200)

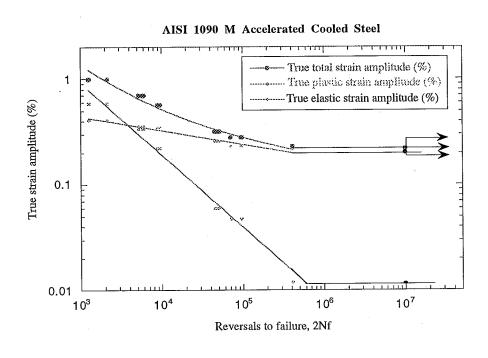


Fig. 4 Constant amplitude fully reversed strain-life curve for AISI 1090M-Accelerated Cooled steel.

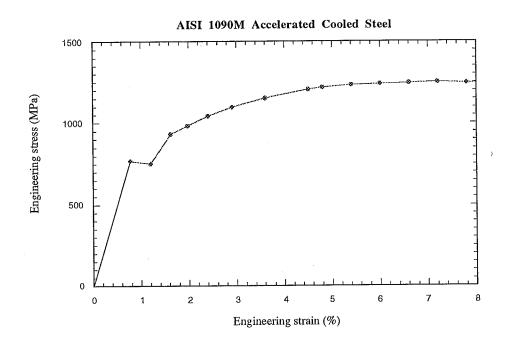


Fig. 5 Monotonic stress-strain curve for AISI 1090M-Accelerated Cooled steel

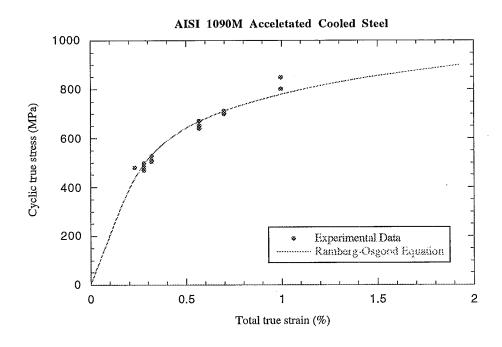


Fig. 6 Cyclic stress-strain curve for AISI 1090M-Accelerated Cooled steel

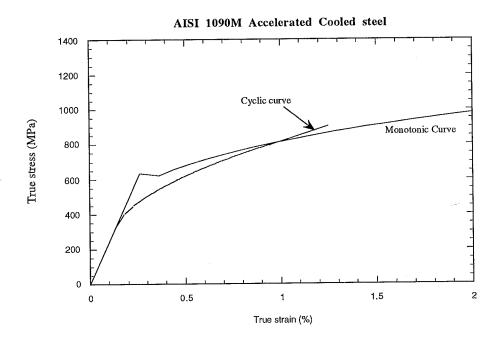


Fig. 7 Monotonic and Cyclic stress-strain curves for AISI 1090M-Accelerated Cooled steel

Table 1 Chemical composition of AISI 1090M-Accelerated Cooled steel

	0.000
Carbon, C	0.88%
Manganese, Mn	0.58%
Phosphorous, P	0.002%
Sulfur, S	0.021%
Silicon, Si	0.22%
Copper, Cu	0.28%
Nickel, Ni	0.08%
Chromium, Cr	0.14%
Molybdenum, Mo	0.02%
Vanadium, Va	0.030%
Calcium, Ca	0.002%
Boron, Bo	0.0024%
Aluminum, Al	<0.001%
Titanium, Ti	0.001%
Oxygen, O	0.0024%
Columbium, Cb	0.003%

TRUE TRUE TRUE TRUE TOtal Strain Suress Amplitude (%) O.997 742.75 O.654 O.998 673.61 O.384 O.698 673.61 O.387 O.569 673.61 O.270 O.569 674.71 O.270 O.569 673.61 O.279 O.569 644.71 O.279 O.569 614.52 O.284 O.369 550.89 O.113 O.369 550.89 O.113 O.369 550.89 O.113 O.369 539.87 O.068 O.319 496.85 O.088 O.279 505.25 O.044 O.259 440.66 O.054	Table 2 Tensile and Fatigue Test Data for 1090M (GM) Hot Form-Accelerated Cooled steel		TRUE (50% load drop) MONOTONIC	Elastic Strain Fatigue Life Young's Hardness	f) Mc	0.342 3144.0 217.27 28		0.346 - 2766.0 219.27	0.314 7394.0 214.5 27.50	0.311 ———————————————————————————————————	0.328 — 7000.0 214.5	13766	0.289 11546 219.5	0.284 13350 219.5 28	52700 206.85	59796	0.246 55296 206.85	0.250 97876 219.50	0.230 77750 219.50 28	0.229 90000 219.50	0.234 272774 219.00	0.215 1940920 219.00	0.201 1.0000e+07* 204.88	0.214 1.0000e+07* 204.88 27.5	0.204 1.0000e+07* 204.88
	Table 2 Tensile and Fatigue	メ	TRUE	Stress Amplifude	(MPa)	742.75	792.55	751.41	680.089	673.61	710.31	644.71	625.30	614.52	548.74	550.89	531.53	539.87	496.85	494.70	505.25	464.33	434.21	462.16	440.66

* Run ou

Appendix 1

Monotonic Properties for AISI 1090M (GM) Hot Form-Accelerated Cooled steel

Average Elastic Modulus, E	=	214.40	GPa
Upper Yield Strength	=	772.70 1	
Lower Yield Strength		759.501	
Ultimate tensile Strength	=	1124.01	
% Elongation	=	18.00	%
% Reduction of Area	=	37.59	
True fracture strain, $Ln(A_i/A_f)$	=		%
True fracture stress, $\sigma_f = \frac{P_f}{A_f}$	=		МРа
Bridgman correction, $\sigma_f = \frac{P_f}{A_f} / \left(1 + \frac{4R}{D_f}\right)$	$Ln\left(1+\frac{D_f}{4R}\right)$	=840 M	1Pa
Monotonic strength coefficient, K	_	1575.5 N	И Ра
Monotonic strain hardening exponent, n	=	0.1083	
Hardness, Rockwell C (HRC)	=	28 HRC	
Hardness, Brinell		272	

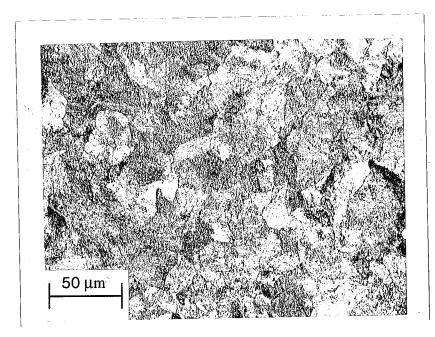
Cyclic Properties for AISI 1090M (GM) Hot Form-Accelerated Cooled steel

= 602 MPa
1964.20 MPa
0.1903
1715.16 MPa
-0.1046
0.7978
-0.5996

 P_f : A_i and A_f : R: Load at fracture.

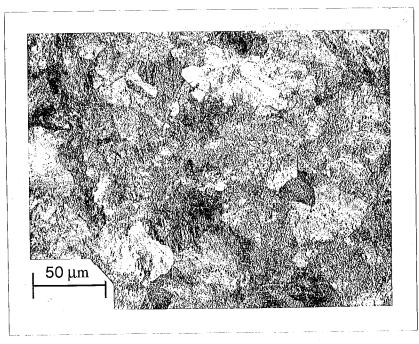
Specimen cross-section area before and after fracture. Specimen neck radius.

Specimen diameter at fracture.



ITER 10: Photomicrograph of SAE 1090 steel, Austempered to Rc-29. 500X Mag.

(a) Longitudinal direction



(b) Transverse direction

Fig. 2 Photomicrographs of AISI 1090 quenched and tempered steel (X500):

(a) Longitudinal direction, and (b) Transverse direction.

AISI Bar Application Group Fatigue Project Test Matrix (6-9-98)

Bar or Part	HR Bar HR Bar	Hr Bar Axles Axles	Hr Bar Stab Bar Stab Bar Stab Bar Stab Bar	Hr Bar Hr Bar	Hr Bar Hr Bar	Hr Bar Hr Bar Hr Bar	Hr Bar Hr Bar Hr Bar	Hr Bar Hr Bar
School	Waterloo Waterloo	Waterloo Waterloo Waterloo	Waterloo Waterloo Waterloo Waterloo Waterloo	Toledo Toledo	Toledo Toledo	Toledo Toledo Toledo	Toledo Toledo Toledo	Toledo
Microstrücture	Ferrite/Pearlite Ferrite/Pearlite	Ferrite/Pearlite Martensite Martensite	Pearlite Martensite/Bainite Martensite Martensite/Bainite Bainite	Ferrite/Pearlite Martensite	Ferrite/Pearlite Martensite	Ferrite/Pearlite Martensite Ferrite/Pearlite	Ferrite/Pearlite Ferrite/Pearlite Martensite	Ferrite/Pearlite Ferrite/Pearlite
Fabricated Hdns	Rb 85 min.	Rc 35 max. Rc 58 min.	BHN 341-444 BHN 302-363 BHN 302-363 BHN 3.0-3.5	BHN 229-269	BHN 229-269	BHN 229-269 TBD	Rb 85 min. Rc 20-30	RC 25-30
Part Fabrication	Normalize 1650F Cold Size/Form	Normalize 1650F Hot Forge, Cold Extrude (Core) Induction Surface Hardened (Case)	Normalize 1650F Hot Form + Acc. Cool Hot Form + Q&T Hot Form + Acc. Cool Hot Form + Austemper	Normalize 1650F Reheat, Q&T	Normalize 1650F Reheat, Q&T	Normalize 1650F Reheat, Q&T Normalize @ High Temp (1750F)	Normalize 1650F Cold Size/Form Reheat, Q&T, (Temper @ 930F)	Normalized 1650F Reheat, 2250F, Deform @ 2000F, FAC
Status	Complete 2-4-98 Complete 2-4-98	Complete 7-30-97 Complete 6-9-98 Complete 6-9-98	Complete 6-9-98 Complete 6-9-98 Complete 2-4-98 Complete 2-4-98 Complete 6-9-98	Complete 6-9-98 Complete 6-9-98	Complete 6-9/98 Complete 6-9-98	@Toledo @Toledo @Toledo	Complete 10-9-97 Complete 6-9-98 @Toledo-Re-H.T.	@Toledo @Toledo
Supplier	RES RES	Stelco Chrysler Chrysler	NS Chrysler Chrysler GM/AMM	RES	S S S	S S S	Stelco Stelco Stelco	Inland Inland
Steel	1541 1541	1050M 1050M 1050M	1090 1090M 1090 1090M 1090	1141(AIFG) 1141(AIFG)	1141(NbFG) 1141(NbFG)	1141(VFG) 1141(VFG) 1141(VFG)	1038 1038 1038	10V45 10V45
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*These iterations will be supplied by GM, and funded separately from the rest of the program.

