

## Results for filtExample\_2.5 : Crack Propagation Plate Edge Flaw

Author: edit this in file makereport5

Affiliation:

Sat Nov 2 20:59:46 EDT 2013

Simulation input data:

**B**= 10.0 mm

**W**= 70.0 mm

**a<sub>0</sub>**= 1.5 mm

#MATERIAL= merged\_a36\_fitted.html

#TYPE= plate\_edge\_flaw

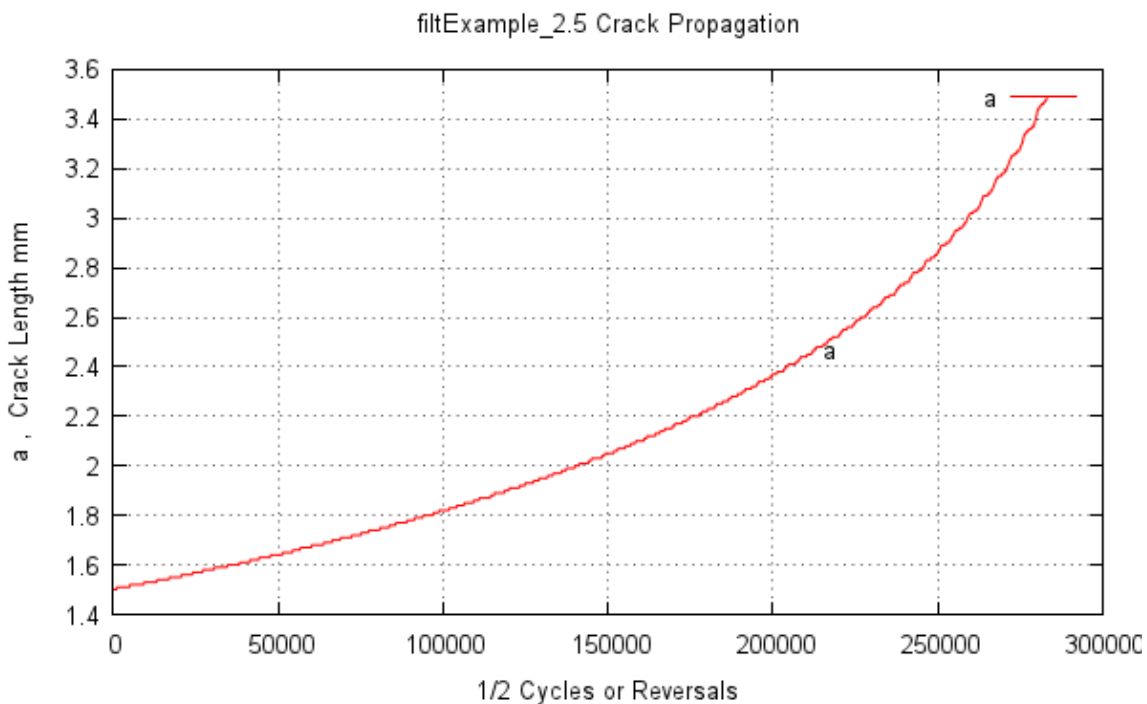
#ACTIVATE\_MmMb= 1

M=Mkm=Mkb=fw=1.0

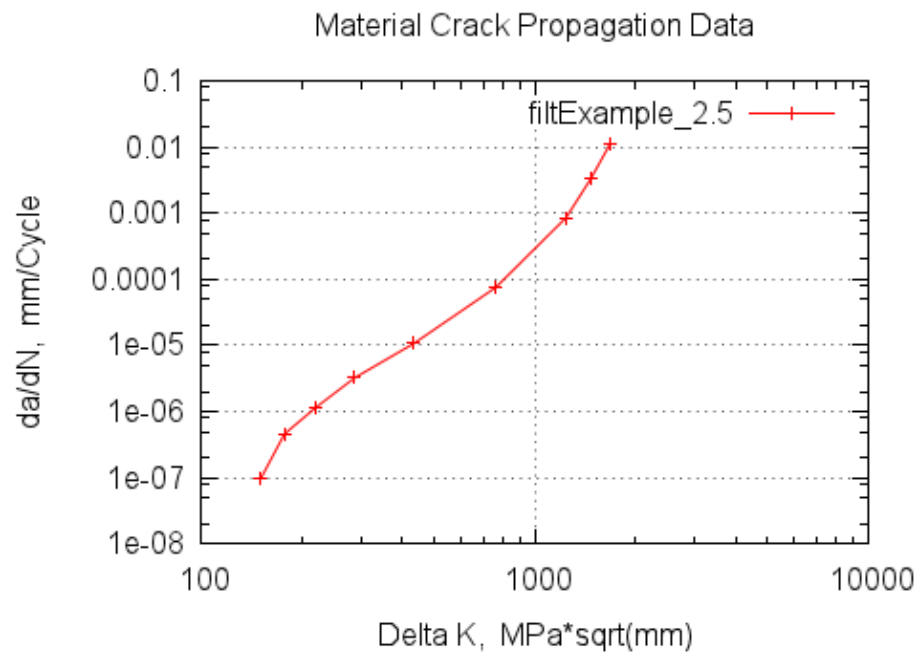
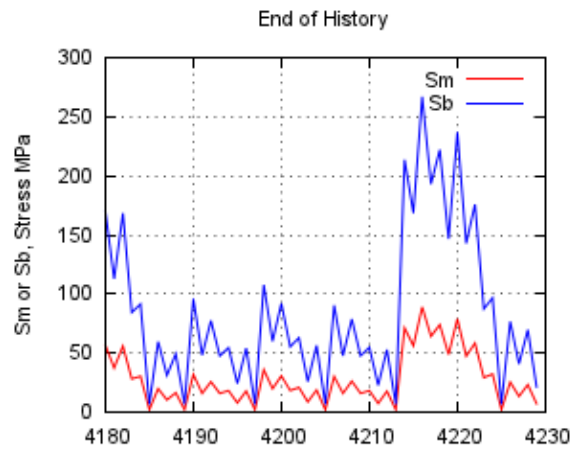
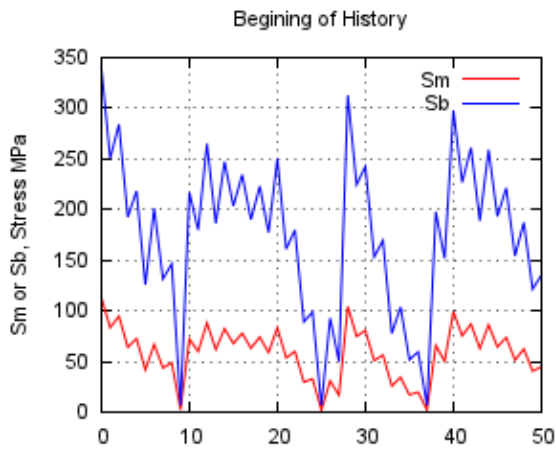
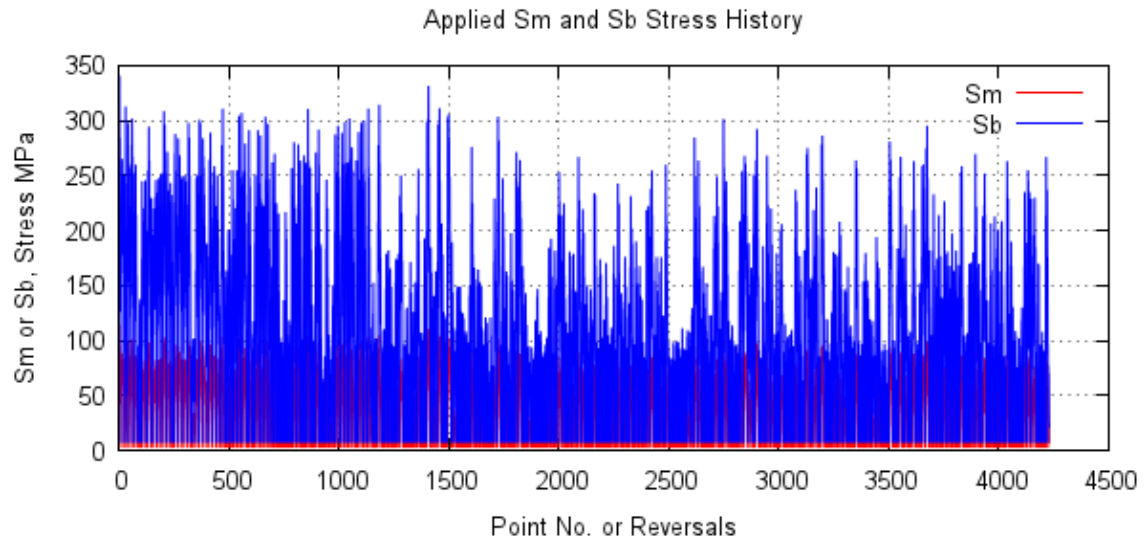
Crack Propagation Results:

( # plateEdgeFlaw.f vers. 3.10 # makereport1 vers. 2.1 )

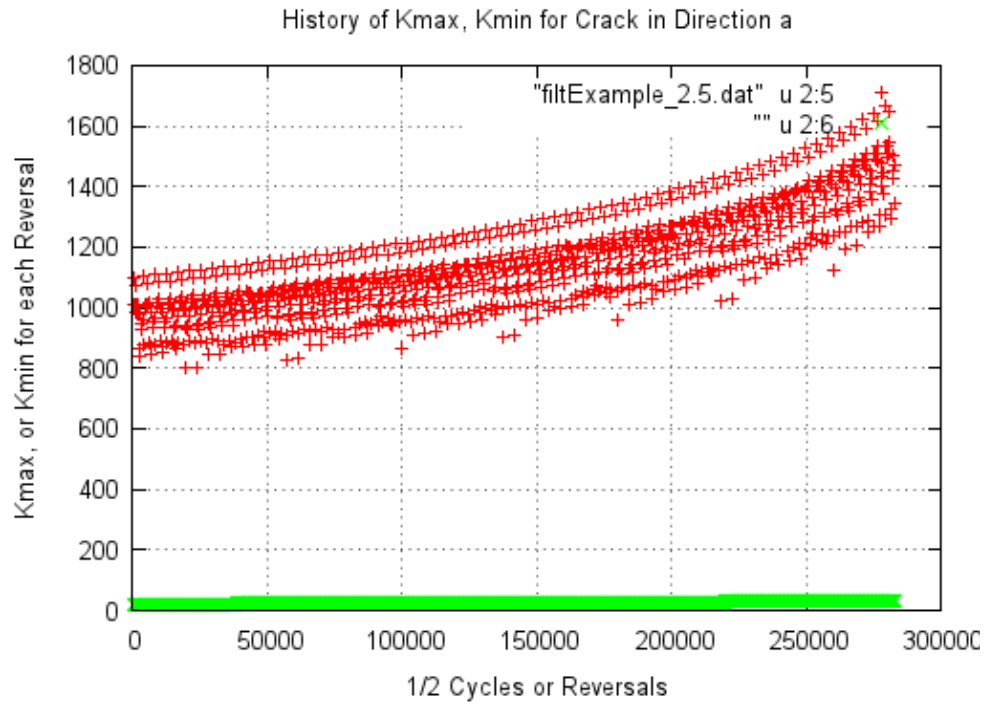
- No. of Reversals= 283148 revs. or 141574 cycles
- Final \_\_\_\_\_ **a** = 0.348E+01 mm
- No. of History Reprs.= 68 reps. + 1 revs.



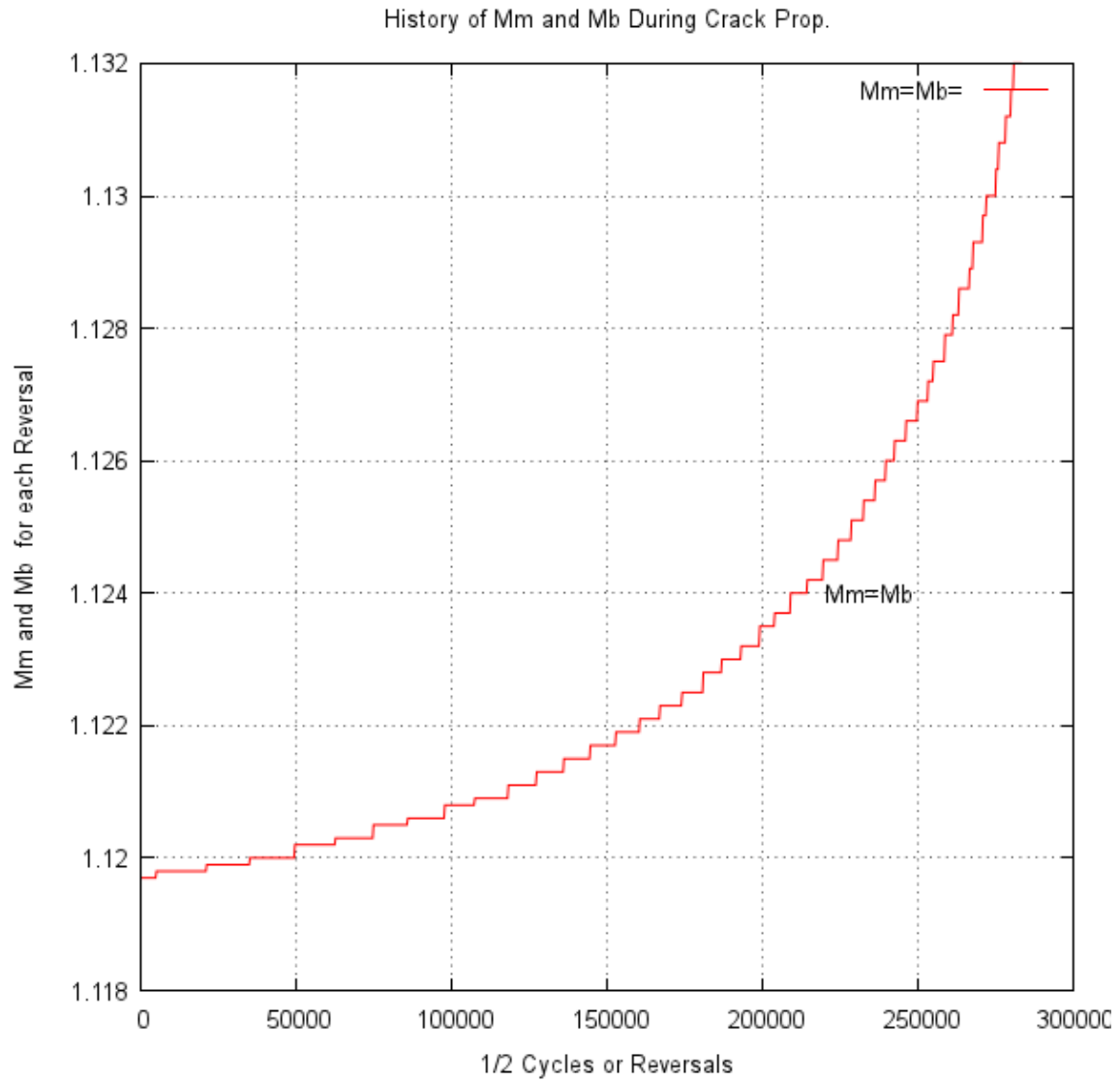
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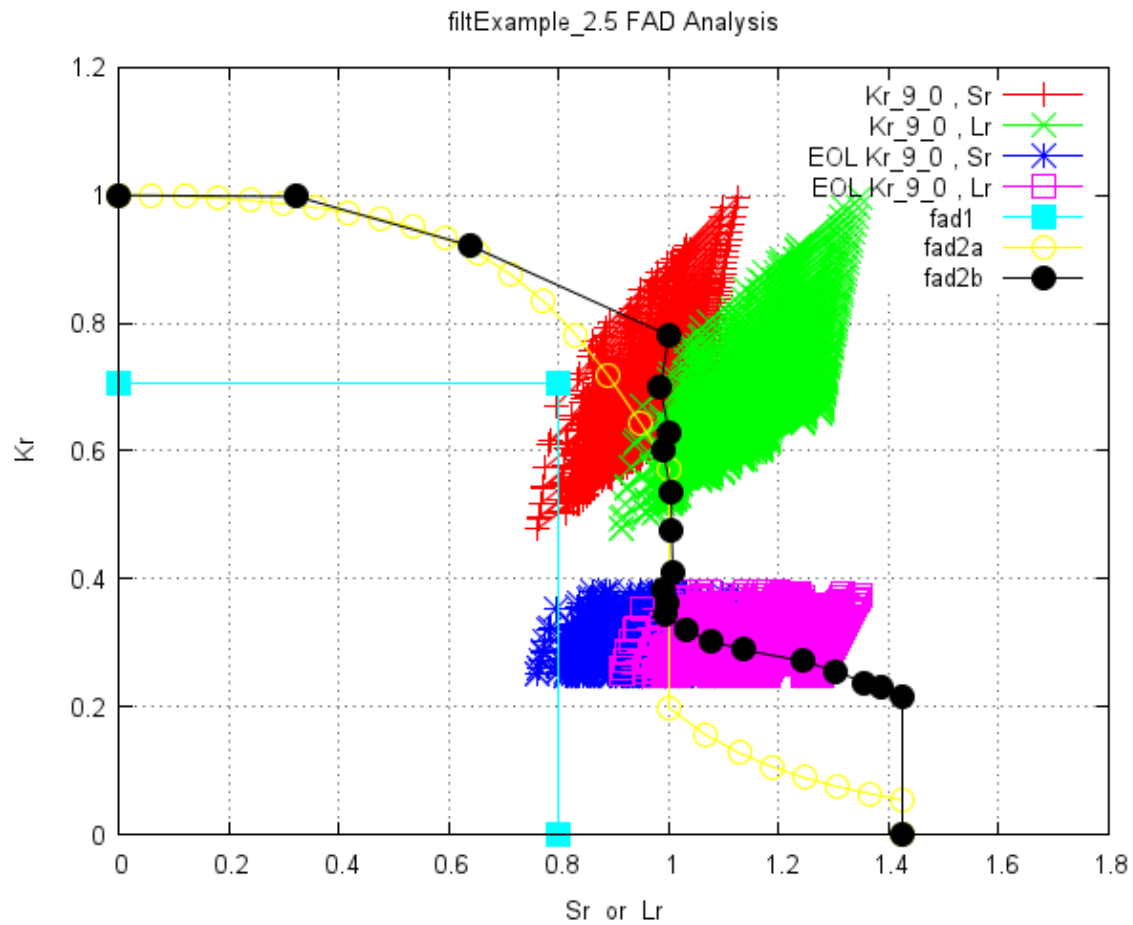
## FAD Results for filtExample\_2.5

#TensileFile= a36\_Mattos\_mono\_engrSS\_FLAT.txt

#PmEOL= 70. #PbEOL= 100.

#Kmat= 1675.

# plateEdgeFAD.f vers. 0.5



## Crack Initiation Life Results for filtExample\_2.5 (Assume $K_t = 1.8$ for welds)

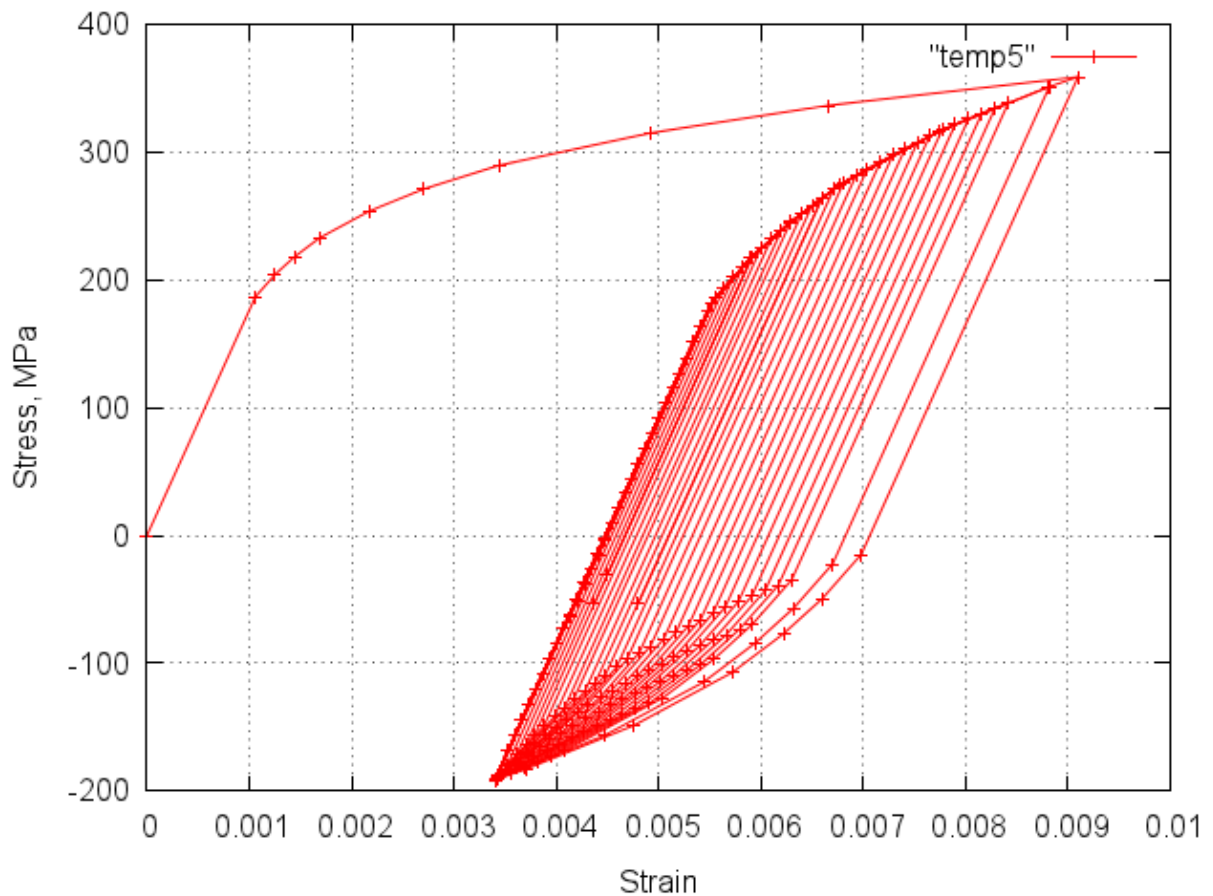
Files Used:

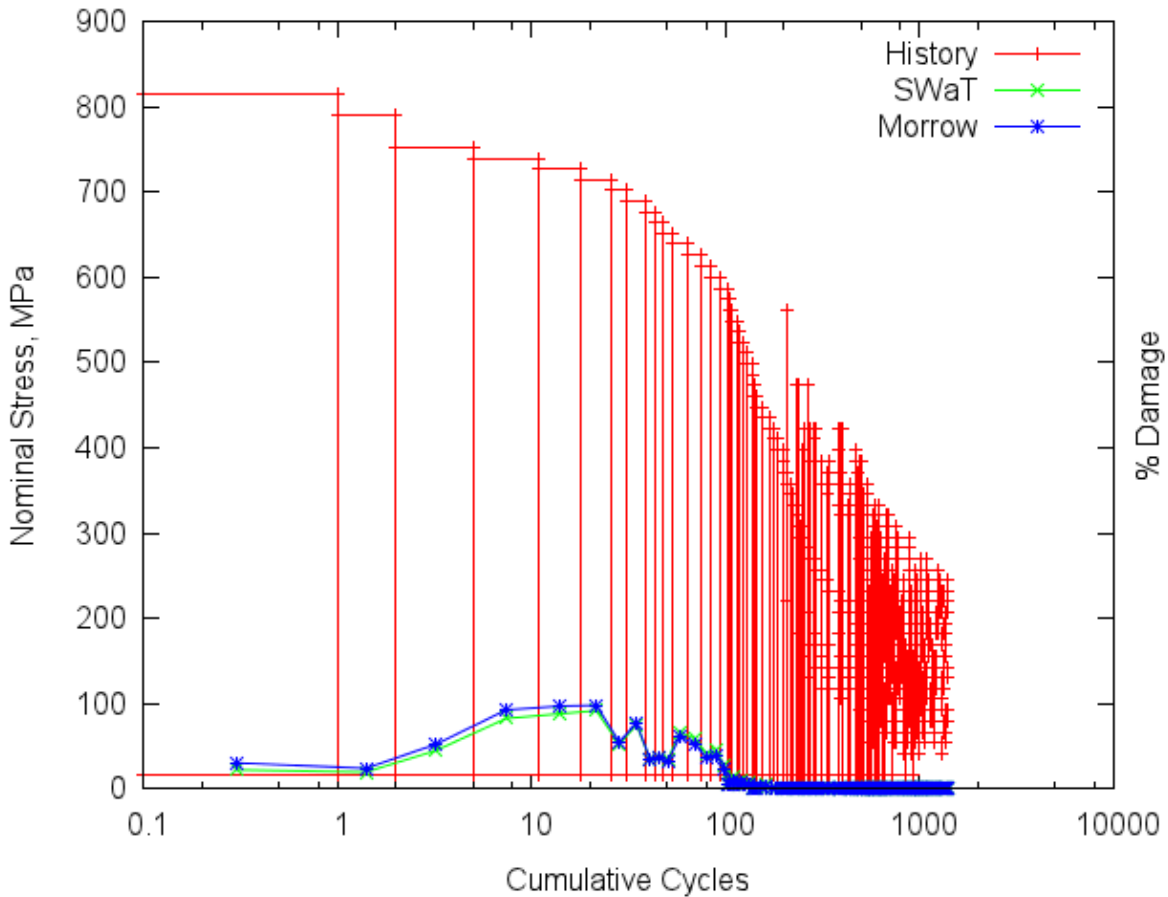
- [Stress History \(Sb+Sm\)](#)
- [Rainflow File](#)
- [Material File](#)

Predicted History Repetitions to Initiation:

StrainLife_Reps	SWaT_Life_Reps	StressLife_Reps	Morrow_Reps	Goodman_Reps	(Reps= Repetitions)
999.3	594.5	999.3	438.7	234.6	

### Local Stress and Strain Response:



**Cumulative Cycle Plot of History and Damage:**

(Rectangles are Rainflow Cycle Sets: Sorted by Range: largest on Left)

**Detailed Damage for each Rainflow Cycle Set:**

Loop	Smax	Smin	N	Sigmax	Sigmin	Delta	Epsmax	Epsmin	DeltaEps	%Eps	%SWaT	%Sts	%Morrow
1	815.4	16.2	1.0	359.	-191.	550.	0.00910	0.00340	0.00570	2.3	2.5	2.3	3.4
2	790.2	16.2	1.0	351.	-191.	543.	0.00882	0.00340	0.00542	2.0	2.2	2.0	2.7
3	752.4	16.2	3.0	339.	-191.	530.	0.00842	0.00340	0.00502	4.8	5.0	4.8	5.8
4	739.8	16.2	6.0	335.	-191.	526.	0.00829	0.00340	0.00489	8.8	9.2	8.8	10.3
5	727.2	16.2	7.0	331.	-191.	522.	0.00816	0.00340	0.00476	9.4	9.8	9.4	10.8
6	714.6	16.2	8.0	326.	-191.	517.	0.00803	0.00340	0.00463	9.8	10.2	9.8	10.9
7	702.0	16.2	5.0	322.	-191.	513.	0.00790	0.00340	0.00450	5.6	5.8	5.6	6.1
8	689.4	16.2	8.0	317.	-191.	509.	0.00777	0.00340	0.00438	8.2	8.4	8.2	8.6
9	676.8	16.2	4.0	313.	-191.	504.	0.00765	0.00340	0.00425	3.7	3.8	3.7	3.8
10	664.2	16.2	5.0	308.	-191.	499.	0.00753	0.00340	0.00413	4.2	4.2	4.2	4.1
11	651.6	16.2	5.0	303.	-191.	494.	0.00741	0.00340	0.00401	3.8	3.8	3.8	3.6
12	639.0	16.2	11.0	298.	-191.	489.	0.00729	0.00340	0.00390	7.4	7.4	7.4	6.8
13	626.4	16.2	11.0	293.	-191.	484.	0.00717	0.00340	0.00378	6.6	6.6	6.6	5.9
14	612.0	16.2	9.0	287.	-191.	478.	0.00704	0.00340	0.00365	4.7	4.7	4.7	4.1
15	599.4	16.2	11.0	282.	-191.	473.	0.00693	0.00340	0.00353	5.2	5.0	5.2	4.3
16	586.8	16.2	8.0	277.	-191.	468.	0.00681	0.00340	0.00342	3.3	3.1	3.3	2.7
17	574.2	16.2	2.0	271.	-191.	462.	0.00671	0.00340	0.00331	0.7	0.7	0.7	0.6
18	561.6	16.2	3.0	265.	-191.	456.	0.00660	0.00340	0.00321	1.0	0.9	1.0	0.7
19	549.0	16.2	6.0	258.	-191.	450.	0.00650	0.00340	0.00310	1.7	1.5	1.7	1.1
20	536.4	16.2	4.0	252.	-191.	443.	0.00640	0.00340	0.00300	1.0	0.8	1.0	0.6
21	523.8	16.2	6.0	246.	-191.	437.	0.00629	0.00340	0.00290	1.2	1.0	1.2	0.8

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22	511.2	16.2	5.0	239.	-191.	430.	0.00620	0.00340	0.00280	0.9	0.7	0.9	0.5
23	498.6	16.2	8.0	232.	-191.	423.	0.00610	0.00340	0.00270	1.2	0.9	1.2	0.7
24	486.0	16.2	2.0	225.	-191.	416.	0.00600	0.00340	0.00261	0.2	0.2	0.2	0.1
25	473.4	16.2	2.0	218.	-191.	409.	0.00591	0.00340	0.00251	0.2	0.1	0.2	0.1
26	460.8	16.2	5.0	210.	-191.	401.	0.00582	0.00340	0.00242	0.4	0.3	0.4	0.2
27	448.2	16.2	8.0	202.	-191.	393.	0.00573	0.00340	0.00233	0.5	0.4	0.5	0.3
28	435.6	16.2	15.0	194.	-191.	385.	0.00564	0.00340	0.00224	0.8	0.5	0.8	0.4
29	423.0	16.2	10.0	186.	-191.	377.	0.00555	0.00340	0.00216	0.4	0.2	0.4	0.2
30	410.4	16.2	7.0	176.	-191.	367.	0.00548	0.00340	0.00208	0.0	0.0	0.0	0.0
31	397.8	16.2	14.0	164.	-191.	355.	0.00541	0.00340	0.00201	0.0	0.0	0.0	0.0
32	385.2	16.2	1.0	152.	-191.	343.	0.00534	0.00340	0.00195	0.0	0.0	0.0	0.0
33	370.8	16.2	7.0	139.	-191.	330.	0.00527	0.00340	0.00187	0.0	0.0	0.0	0.0
34	561.6	219.6	1.0	265.	-54.	318.	0.00660	0.00480	0.00181	0.0	0.0	0.0	0.0
35	358.2	16.2	8.0	127.	-191.	318.	0.00520	0.00340	0.00181	0.0	0.0	0.0	0.0
36	345.6	16.2	8.0	115.	-191.	307.	0.00514	0.00340	0.00174	0.0	0.0	0.0	0.0
37	333.0	16.2	6.0	104.	-191.	295.	0.00507	0.00340	0.00167	0.0	0.0	0.0	0.0
38	320.4	16.2	4.0	92.	-191.	283.	0.00500	0.00340	0.00161	0.0	0.0	0.0	0.0
39	473.4	181.8	1.0	218.	-54.	271.	0.00591	0.00437	0.00154	0.0	0.0	0.0	0.0
40	307.8	16.2	7.0	80.	-191.	271.	0.00494	0.00340	0.00154	0.0	0.0	0.0	0.0
41	295.2	16.2	9.0	68.	-191.	260.	0.00487	0.00340	0.00147	0.0	0.0	0.0	0.0
42	397.8	130.5	2.0	164.	-85.	249.	0.00541	0.00400	0.00141	0.0	0.0	0.0	0.0
43	423.0	155.9	2.0	186.	-63.	249.	0.00555	0.00414	0.00141	0.0	0.0	0.0	0.0
44	282.6	16.2	12.0	57.	-191.	248.	0.00480	0.00340	0.00141	0.0	0.0	0.0	0.0
45	473.4	207.0	1.0	218.	-30.	248.	0.00591	0.00450	0.00141	0.0	0.0	0.0	0.0
46	385.2	130.5	3.0	152.	-85.	237.	0.00534	0.00400	0.00134	0.0	0.0	0.0	0.0
47	423.0	168.5	1.0	186.	-51.	237.	0.00555	0.00421	0.00134	0.0	0.0	0.0	0.0
48	270.0	16.2	13.0	45.	-191.	236.	0.00474	0.00340	0.00134	0.0	0.0	0.0	0.0
49	385.2	143.1	2.0	152.	-73.	225.	0.00534	0.00407	0.00128	0.0	0.0	0.0	0.0
50	410.4	168.5	1.0	176.	-50.	225.	0.00548	0.00420	0.00128	0.0	0.0	0.0	0.0
51	423.0	181.8	2.0	186.	-39.	224.	0.00555	0.00428	0.00127	0.0	0.0	0.0	0.0
52	257.4	16.2	22.0	33.	-191.	224.	0.00467	0.00340	0.00127	0.0	0.0	0.0	0.0
53	358.2	117.7	1.0	127.	-97.	224.	0.00520	0.00393	0.00127	0.0	0.0	0.0	0.0
54	385.2	155.9	1.0	152.	-61.	213.	0.00534	0.00413	0.00121	0.0	0.0	0.0	0.0
55	244.8	16.2	17.0	21.	-191.	213.	0.00460	0.00340	0.00121	0.0	0.0	0.0	0.0
56	345.6	117.7	3.0	115.	-97.	212.	0.00514	0.00393	0.00120	0.0	0.0	0.0	0.0
57	370.8	143.1	3.0	139.	-73.	212.	0.00527	0.00407	0.00120	0.0	0.0	0.0	0.0
58	358.2	130.5	5.0	127.	-85.	212.	0.00520	0.00400	0.00120	0.0	0.0	0.0	0.0
59	385.2	168.5	2.0	152.	-50.	202.	0.00534	0.00420	0.00114	0.0	0.0	0.0	0.0
60	232.2	16.2	41.0	10.	-191.	201.	0.00454	0.00340	0.00114	0.0	0.0	0.0	0.0
61	423.0	207.0	1.0	186.	-15.	201.	0.00555	0.00441	0.00114	0.0	0.0	0.0	0.0
62	397.8	181.8	1.0	164.	-37.	201.	0.00541	0.00427	0.00114	0.0	0.0	0.0	0.0
63	333.0	117.7	1.0	104.	-97.	200.	0.00507	0.00393	0.00114	0.0	0.0	0.0	0.0
64	320.4	105.1	1.0	92.	-108.	200.	0.00500	0.00387	0.00114	0.0	0.0	0.0	0.0
65	358.2	143.1	4.0	127.	-73.	200.	0.00520	0.00407	0.00114	0.0	0.0	0.0	0.0
66	370.8	155.9	3.0	139.	-61.	200.	0.00527	0.00413	0.00113	0.0	0.0	0.0	0.0
67	423.0	219.6	1.0	186.	-3.	189.	0.00555	0.00448	0.00107	0.0	0.0	0.0	0.0
68	219.6	16.2	31.0	-2.	-191.	189.	0.00447	0.00340	0.00107	0.0	0.0	0.0	0.0
69	320.4	117.7	2.0	92.	-97.	189.	0.00500	0.00393	0.00107	0.0	0.0	0.0	0.0
70	333.0	130.5	5.0	104.	-85.	188.	0.00507	0.00400	0.00107	0.0	0.0	0.0	0.0
71	345.6	143.1	2.0	115.	-73.	188.	0.00514	0.00407	0.00107	0.0	0.0	0.0	0.0
72	358.2	155.9	2.0	127.	-61.	188.	0.00520	0.00413	0.00107	0.0	0.0	0.0	0.0
73	207.0	16.2	29.0	-14.	-191.	178.	0.00440	0.00340	0.00101	0.0	0.0	0.0	0.0
74	397.8	207.0	1.0	164.	-14.	178.	0.00541	0.00440	0.00101	0.0	0.0	0.0	0.0
75	385.2	194.4	2.0	152.	-25.	178.	0.00534	0.00434	0.00101	0.0	0.0	0.0	0.0
76	333.0	143.1	2.0	104.	-73.	177.	0.00507	0.00407	0.00100	0.0	0.0	0.0	0.0
77	320.4	130.5	3.0	92.	-85.	177.	0.00500	0.00400	0.00100	0.0	0.0	0.0	0.0
78	358.2	168.5	1.0	127.	-50.	177.	0.00520	0.00420	0.00100	0.0	0.0	0.0	0.0
79	370.8	181.8	1.0	139.	-37.	176.	0.00527	0.00427	0.00100	0.0	0.0	0.0	0.0
80	194.4	16.2	15.0	-25.	-191.	166.	0.00434	0.00340	0.00094	0.0	0.0	0.0	0.0
81	270.0	92.3	1.0	45.	-120.	165.	0.00474	0.00380	0.00094	0.0	0.0	0.0	0.0
82	295.2	117.7	3.0	68.	-97.	165.	0.00487	0.00393	0.00094	0.0	0.0	0.0	0.0
83	320.4	143.1	1.0	92.	-73.	165.	0.00500	0.00407	0.00094	0.0	0.0	0.0	0.0



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84	385.2	219.6	1.0	152.	-2.	154.	0.00534	0.00447	0.00087	0.0	0.0	0.0	0.0
85	181.8	16.2	10.0	-37.	-191.	154.	0.00427	0.00340	0.00087	0.0	0.0	0.0	0.0
86	244.8	79.7	1.0	21.	-132.	154.	0.00460	0.00373	0.00087	0.0	0.0	0.0	0.0
87	282.6	117.7	2.0	57.	-97.	153.	0.00480	0.00393	0.00087	0.0	0.0	0.0	0.0
88	295.2	130.5	1.0	68.	-85.	153.	0.00487	0.00400	0.00087	0.0	0.0	0.0	0.0
89	333.0	168.5	1.0	104.	-50.	153.	0.00507	0.00420	0.00087	0.0	0.0	0.0	0.0
90	345.6	181.8	3.0	115.	-37.	152.	0.00514	0.00427	0.00086	0.0	0.0	0.0	0.0
91	207.0	54.4	1.0	-14.	-156.	142.	0.00440	0.00360	0.00081	0.0	0.0	0.0	0.0
92	244.8	92.3	3.0	21.	-120.	142.	0.00460	0.00380	0.00080	0.0	0.0	0.0	0.0
93	168.5	16.2	16.0	-50.	-191.	142.	0.00420	0.00340	0.00080	0.0	0.0	0.0	0.0
94	257.4	105.1	2.0	33.	-108.	142.	0.00467	0.00387	0.00080	0.0	0.0	0.0	0.0
95	333.0	181.8	2.0	104.	-37.	141.	0.00507	0.00427	0.00080	0.0	0.0	0.0	0.0
96	358.2	207.0	1.0	127.	-14.	141.	0.00520	0.00440	0.00080	0.0	0.0	0.0	0.0
97	345.6	194.4	1.0	115.	-25.	141.	0.00514	0.00434	0.00080	0.0	0.0	0.0	0.0
98	232.2	92.3	4.0	10.	-120.	130.	0.00454	0.00380	0.00074	0.0	0.0	0.0	0.0
99	155.9	16.2	18.0	-61.	-191.	130.	0.00413	0.00340	0.00074	0.0	0.0	0.0	0.0
100	244.8	105.1	2.0	21.	-108.	130.	0.00460	0.00387	0.00074	0.0	0.0	0.0	0.0
101	282.6	143.1	1.0	57.	-73.	130.	0.00480	0.00407	0.00074	0.0	0.0	0.0	0.0
102	270.0	130.5	1.0	45.	-85.	130.	0.00474	0.00400	0.00074	0.0	0.0	0.0	0.0
103	295.2	155.9	2.0	68.	-61.	130.	0.00487	0.00413	0.00074	0.0	0.0	0.0	0.0
104	320.4	181.8	3.0	92.	-37.	129.	0.00500	0.00427	0.00073	0.0	0.0	0.0	0.0
105	194.4	67.0	1.0	-25.	-144.	119.	0.00434	0.00366	0.00067	0.0	0.0	0.0	0.0
106	181.8	54.4	1.0	-37.	-156.	119.	0.00427	0.00360	0.00067	0.0	0.0	0.0	0.0
107	207.0	79.7	2.0	-14.	-132.	118.	0.00440	0.00373	0.00067	0.0	0.0	0.0	0.0
108	219.6	92.3	3.0	-2.	-120.	118.	0.00447	0.00380	0.00067	0.0	0.0	0.0	0.0
109	232.2	105.1	2.0	10.	-108.	118.	0.00454	0.00387	0.00067	0.0	0.0	0.0	0.0
110	244.8	117.7	1.0	21.	-97.	118.	0.00460	0.00393	0.00067	0.0	0.0	0.0	0.0
111	143.1	16.2	7.0	-73.	-191.	118.	0.00407	0.00340	0.00067	0.0	0.0	0.0	0.0
112	282.6	155.9	1.0	57.	-61.	118.	0.00480	0.00413	0.00067	0.0	0.0	0.0	0.0
113	295.2	168.5	1.0	68.	-50.	118.	0.00487	0.00420	0.00067	0.0	0.0	0.0	0.0
114	333.0	207.0	2.0	104.	-14.	117.	0.00507	0.00440	0.00067	0.0	0.0	0.0	0.0
115	320.4	194.4	1.0	92.	-25.	117.	0.00500	0.00434	0.00067	0.0	0.0	0.0	0.0
116	307.8	181.8	3.0	80.	-37.	117.	0.00494	0.00427	0.00067	0.0	0.0	0.0	0.0
117	181.8	67.0	3.0	-37.	-144.	107.	0.00427	0.00366	0.00061	0.0	0.0	0.0	0.0
118	194.4	79.7	4.0	-25.	-132.	107.	0.00434	0.00373	0.00061	0.0	0.0	0.0	0.0
119	232.2	117.7	2.0	10.	-97.	107.	0.00454	0.00393	0.00060	0.0	0.0	0.0	0.0
120	130.5	16.2	5.0	-85.	-191.	106.	0.00400	0.00340	0.00060	0.0	0.0	0.0	0.0
121	244.8	130.5	1.0	21.	-85.	106.	0.00460	0.00400	0.00060	0.0	0.0	0.0	0.0
122	282.6	168.5	3.0	57.	-50.	106.	0.00480	0.00420	0.00060	0.0	0.0	0.0	0.0
123	270.0	155.9	1.0	45.	-61.	106.	0.00474	0.00413	0.00060	0.0	0.0	0.0	0.0
124	333.0	219.6	3.0	104.	-2.	106.	0.00507	0.00447	0.00060	0.0	0.0	0.0	0.0
125	320.4	207.0	3.0	92.	-14.	106.	0.00500	0.00440	0.00060	0.0	0.0	0.0	0.0
126	307.8	194.4	1.0	80.	-25.	106.	0.00494	0.00434	0.00060	0.0	0.0	0.0	0.0
127	295.2	181.8	4.0	68.	-37.	106.	0.00487	0.00427	0.00060	0.0	0.0	0.0	0.0
128	194.4	92.3	2.0	-25.	-120.	95.	0.00434	0.00380	0.00054	0.0	0.0	0.0	0.0
129	181.8	79.7	4.0	-37.	-132.	95.	0.00427	0.00373	0.00054	0.0	0.0	0.0	0.0
130	207.0	105.1	1.0	-14.	-108.	95.	0.00440	0.00387	0.00054	0.0	0.0	0.0	0.0
131	219.6	117.7	5.0	-2.	-97.	95.	0.00447	0.00393	0.00054	0.0	0.0	0.0	0.0
132	244.8	143.1	5.0	21.	-73.	95.	0.00460	0.00407	0.00054	0.0	0.0	0.0	0.0
133	232.2	130.5	7.0	10.	-85.	95.	0.00454	0.00400	0.00054	0.0	0.0	0.0	0.0
134	117.7	16.2	6.0	-97.	-191.	94.	0.00393	0.00340	0.00054	0.0	0.0	0.0	0.0
135	270.0	168.5	6.0	45.	-50.	94.	0.00474	0.00420	0.00054	0.0	0.0	0.0	0.0
136	257.4	155.9	4.0	33.	-61.	94.	0.00467	0.00413	0.00054	0.0	0.0	0.0	0.0
137	168.5	67.0	7.0	-50.	-144.	94.	0.00420	0.00366	0.00054	0.0	0.0	0.0	0.0
138	282.6	181.8	4.0	57.	-37.	94.	0.00480	0.00427	0.00053	0.0	0.0	0.0	0.0
139	320.4	219.6	7.0	92.	-2.	94.	0.00500	0.00447	0.00053	0.0	0.0	0.0	0.0
140	307.8	207.0	3.0	80.	-14.	94.	0.00494	0.00440	0.00053	0.0	0.0	0.0	0.0
141	207.0	117.7	5.0	-14.	-97.	83.	0.00440	0.00393	0.00047	0.0	0.0	0.0	0.0
142	194.4	105.1	7.0	-25.	-108.	83.	0.00434	0.00387	0.00047	0.0	0.0	0.0	0.0
143	232.2	143.1	7.0	10.	-73.	83.	0.00454	0.00407	0.00047	0.0	0.0	0.0	0.0
144	219.6	130.5	8.0	-2.	-85.	83.	0.00447	0.00400	0.00047	0.0	0.0	0.0	0.0
145	105.1	16.2	8.0	-108.	-191.	83.	0.00387	0.00340	0.00047	0.0	0.0	0.0	0.0

# Results for filtExample\_2.5 : Crack Propagation Plate Edge Flaw

146	257.4	168.5	5.0	33.	-50.	83.	0.00467	0.00420	0.00047	0.0	0.0	0.0	0.
147	244.8	155.9	3.0	21.	-61.	83.	0.00460	0.00413	0.00047	0.0	0.0	0.0	0.
148	155.9	67.0	12.0	-61.	-144.	83.	0.00413	0.00366	0.00047	0.0	0.0	0.0	0.
149	168.5	79.7	5.0	-50.	-132.	83.	0.00420	0.00373	0.00047	0.0	0.0	0.0	0.
150	143.1	54.4	8.0	-73.	-156.	83.	0.00407	0.00360	0.00047	0.0	0.0	0.0	0.
151	282.6	194.4	1.0	57.	-25.	82.	0.00480	0.00434	0.00047	0.0	0.0	0.0	0.
152	270.0	181.8	3.0	45.	-37.	82.	0.00474	0.00427	0.00047	0.0	0.0	0.0	0.
153	307.8	219.6	3.0	80.	-2.	82.	0.00494	0.00447	0.00047	0.0	0.0	0.0	0.
154	295.2	207.0	6.0	68.	-14.	82.	0.00487	0.00440	0.00047	0.0	0.0	0.0	0.
155	194.4	117.7	16.0	-25.	-97.	71.	0.00434	0.00393	0.00040	0.0	0.0	0.0	0.
156	181.8	105.1	13.0	-37.	-108.	71.	0.00427	0.00387	0.00040	0.0	0.0	0.0	0.
157	207.0	130.5	15.0	-14.	-85.	71.	0.00440	0.00400	0.00040	0.0	0.0	0.0	0.
158	219.6	143.1	11.0	-2.	-73.	71.	0.00447	0.00407	0.00040	0.0	0.0	0.0	0.
159	232.2	155.9	7.0	10.	-61.	71.	0.00454	0.00413	0.00040	0.0	0.0	0.0	0.
160	244.8	168.5	4.0	21.	-50.	71.	0.00460	0.00420	0.00040	0.0	0.0	0.0	0.
161	117.7	41.6	1.0	-97.	-168.	71.	0.00393	0.00353	0.00040	0.0	0.0	0.0	0.
162	168.5	92.3	6.0	-50.	-120.	71.	0.00420	0.00380	0.00040	0.0	0.0	0.0	0.
163	92.3	16.2	6.0	-120.	-191.	71.	0.00380	0.00340	0.00040	0.0	0.0	0.0	0.
164	130.5	54.4	13.0	-85.	-156.	71.	0.00400	0.00360	0.00040	0.0	0.0	0.0	0.
165	143.1	67.0	14.0	-73.	-144.	71.	0.00407	0.00366	0.00040	0.0	0.0	0.0	0.
166	155.9	79.7	4.0	-61.	-132.	71.	0.00413	0.00373	0.00040	0.0	0.0	0.0	0.
167	282.6	207.0	2.0	57.	-14.	70.	0.00480	0.00440	0.00040	0.0	0.0	0.0	0.
168	270.0	194.4	1.0	45.	-25.	70.	0.00474	0.00434	0.00040	0.0	0.0	0.0	0.
169	257.4	181.8	1.0	33.	-37.	70.	0.00467	0.00427	0.00040	0.0	0.0	0.0	0.
170	295.2	219.6	1.0	68.	-2.	70.	0.00487	0.00447	0.00040	0.0	0.0	0.0	0.
171	181.8	117.7	14.0	-37.	-97.	60.	0.00427	0.00393	0.00034	0.0	0.0	0.0	0.
172	207.0	143.1	5.0	-14.	-73.	59.	0.00440	0.00407	0.00034	0.0	0.0	0.0	0.
173	194.4	130.5	8.0	-25.	-85.	59.	0.00434	0.00400	0.00034	0.0	0.0	0.0	0.
174	219.6	155.9	4.0	-2.	-61.	59.	0.00447	0.00413	0.00034	0.0	0.0	0.0	0.
175	232.2	168.5	2.0	10.	-50.	59.	0.00454	0.00420	0.00034	0.0	0.0	0.0	0.
176	105.1	41.6	1.0	-108.	-168.	59.	0.00387	0.00353	0.00034	0.0	0.0	0.0	0.
177	130.5	67.0	5.0	-85.	-144.	59.	0.00400	0.00366	0.00034	0.0	0.0	0.0	0.
178	79.7	16.2	2.0	-132.	-191.	59.	0.00373	0.00340	0.00034	0.0	0.0	0.0	0.
179	155.9	92.3	6.0	-61.	-120.	59.	0.00413	0.00380	0.00034	0.0	0.0	0.0	0.
180	117.7	54.4	16.0	-97.	-156.	59.	0.00393	0.00360	0.00033	0.0	0.0	0.0	0.
181	143.1	79.7	3.0	-73.	-132.	59.	0.00407	0.00373	0.00033	0.0	0.0	0.0	0.
182	168.5	105.1	8.0	-50.	-108.	59.	0.00420	0.00387	0.00033	0.0	0.0	0.0	0.
183	257.4	194.4	2.0	33.	-25.	59.	0.00467	0.00434	0.00033	0.0	0.0	0.0	0.
184	244.8	181.8	3.0	21.	-37.	59.	0.00460	0.00427	0.00033	0.0	0.0	0.0	0.
185	181.8	130.5	4.0	-37.	-85.	48.	0.00427	0.00400	0.00027	0.0	0.0	0.0	0.
186	207.0	155.9	2.0	-14.	-61.	48.	0.00440	0.00413	0.00027	0.0	0.0	0.0	0.
187	219.6	168.5	4.0	-2.	-50.	48.	0.00447	0.00420	0.00027	0.0	0.0	0.0	0.
188	117.7	67.0	7.0	-97.	-144.	47.	0.00393	0.00366	0.00027	0.0	0.0	0.0	0.
189	130.5	79.7	8.0	-85.	-132.	47.	0.00400	0.00373	0.00027	0.0	0.0	0.0	0.
190	105.1	54.4	4.0	-108.	-156.	47.	0.00387	0.00360	0.00027	0.0	0.0	0.0	0.
191	92.3	41.6	2.0	-120.	-168.	47.	0.00380	0.00353	0.00027	0.0	0.0	0.0	0.
192	143.1	92.3	11.0	-73.	-120.	47.	0.00407	0.00380	0.00027	0.0	0.0	0.0	0.
193	168.5	117.7	3.0	-50.	-97.	47.	0.00420	0.00393	0.00027	0.0	0.0	0.0	0.
194	155.9	105.1	7.0	-61.	-108.	47.	0.00413	0.00387	0.00027	0.0	0.0	0.0	0.
195	270.0	219.6	3.0	45.	-2.	47.	0.00474	0.00447	0.00027	0.0	0.0	0.0	0.
196	232.2	181.8	1.0	10.	-37.	47.	0.00454	0.00427	0.00027	0.0	0.0	0.0	0.
197	257.4	207.0	3.0	33.	-14.	47.	0.00467	0.00440	0.00027	0.0	0.0	0.0	0.
198	244.8	194.4	2.0	21.	-25.	47.	0.00460	0.00434	0.00027	0.0	0.0	0.0	0.
199	181.8	143.1	3.0	-37.	-73.	36.	0.00427	0.00407	0.00020	0.0	0.0	0.0	0.
200	194.4	155.9	4.0	-25.	-61.	36.	0.00434	0.00413	0.00020	0.0	0.0	0.0	0.
201	207.0	168.5	9.0	-14.	-50.	36.	0.00440	0.00420	0.00020	0.0	0.0	0.0	0.
202	105.1	67.0	9.0	-108.	-144.	36.	0.00387	0.00366	0.00020	0.0	0.0	0.0	0.
203	130.5	92.3	5.0	-85.	-120.	36.	0.00400	0.00380	0.00020	0.0	0.0	0.0	0.
204	155.9	117.7	2.0	-61.	-97.	36.	0.00413	0.00393	0.00020	0.0	0.0	0.0	0.
205	117.7	79.7	15.0	-97.	-132.	35.	0.00393	0.00373	0.00020	0.0	0.0	0.0	0.
206	168.5	130.5	3.0	-50.	-85.	35.	0.00420	0.00400	0.00020	0.0	0.0	0.0	0.
207	92.3	54.4	2.0	-120.	-156.	35.	0.00380	0.00360	0.00020	0.0	0.0	0.0	0.

## Results for filtExample\_2.5 : Crack Propagation Plate Edge Flaw

208	143.1	105.1	6.0	-73.	-108.	35.	0.00407	0.00387	0.00020	0.0	0.0	0.0	0.0
209	270.0	232.2	4.0	45.	10.	35.	0.00474	0.00454	0.00020	0.0	0.0	0.0	0.0
210	219.6	181.8	4.0	-2.	-37.	35.	0.00447	0.00427	0.00020	0.0	0.0	0.0	0.0
211	257.4	219.6	9.0	33.	-2.	35.	0.00467	0.00447	0.00020	0.0	0.0	0.0	0.0
212	232.2	194.4	10.0	10.	-25.	35.	0.00454	0.00434	0.00020	0.0	0.0	0.0	0.0
213	244.8	207.0	4.0	21.	-14.	35.	0.00460	0.00440	0.00020	0.0	0.0	0.0	0.0
214	181.8	155.9	12.0	-37.	-61.	24.	0.00427	0.00413	0.00014	0.0	0.0	0.0	0.0
215	194.4	168.5	19.0	-25.	-50.	24.	0.00434	0.00420	0.00014	0.0	0.0	0.0	0.0
216	117.7	92.3	7.0	-97.	-120.	24.	0.00393	0.00380	0.00013	0.0	0.0	0.0	0.0
217	105.1	79.7	18.0	-108.	-132.	24.	0.00387	0.00373	0.00013	0.0	0.0	0.0	0.0
218	168.5	143.1	10.0	-50.	-73.	24.	0.00420	0.00407	0.00013	0.0	0.0	0.0	0.0
219	130.5	105.1	5.0	-85.	-108.	24.	0.00400	0.00387	0.00013	0.0	0.0	0.0	0.0
220	92.3	67.0	7.0	-120.	-144.	24.	0.00380	0.00366	0.00013	0.0	0.0	0.0	0.0
221	79.7	54.4	6.0	-132.	-156.	24.	0.00373	0.00360	0.00013	0.0	0.0	0.0	0.0
222	155.9	130.5	13.0	-61.	-85.	24.	0.00413	0.00400	0.00013	0.0	0.0	0.0	0.0
223	143.1	117.7	6.0	-73.	-97.	24.	0.00407	0.00393	0.00013	0.0	0.0	0.0	0.0
224	207.0	181.8	21.0	-14.	-37.	23.	0.00440	0.00427	0.00013	0.0	0.0	0.0	0.0
225	257.4	232.2	5.0	33.	10.	23.	0.00467	0.00454	0.00013	0.0	0.0	0.0	0.0
226	219.6	194.4	15.0	-2.	-25.	23.	0.00447	0.00434	0.00013	0.0	0.0	0.0	0.0
227	244.8	219.6	23.0	21.	-2.	23.	0.00460	0.00447	0.00013	0.0	0.0	0.0	0.0
228	232.2	207.0	21.0	10.	-14.	23.	0.00454	0.00440	0.00013	0.0	0.0	0.0	0.0
229	181.8	168.5	9.0	-37.	-50.	12.	0.00427	0.00420	0.00007	0.0	0.0	0.0	0.0
230	105.1	92.3	1.0	-108.	-120.	12.	0.00387	0.00380	0.00007	0.0	0.0	0.0	0.0
231	54.4	41.6	1.0	-156.	-168.	12.	0.00360	0.00353	0.00007	0.0	0.0	0.0	0.0
232	130.5	117.7	12.0	-85.	-97.	12.	0.00400	0.00393	0.00007	0.0	0.0	0.0	0.0
233	79.7	67.0	4.0	-132.	-144.	12.	0.00373	0.00366	0.00007	0.0	0.0	0.0	0.0
234	155.9	143.1	8.0	-61.	-73.	12.	0.00413	0.00407	0.00007	0.0	0.0	0.0	0.0
235	232.2	219.6	9.0	10.	-2.	12.	0.00454	0.00447	0.00007	0.0	0.0	0.0	0.0
236	194.4	181.8	12.0	-25.	-37.	12.	0.00434	0.00427	0.00007	0.0	0.0	0.0	0.0
237	168.5	155.9	11.0	-50.	-61.	12.	0.00420	0.00413	0.00007	0.0	0.0	0.0	0.0
238	207.0	194.4	11.0	-14.	-25.	12.	0.00440	0.00434	0.00007	0.0	0.0	0.0	0.0
239	67.0	54.4	1.0	-144.	-156.	12.	0.00366	0.00360	0.00007	0.0	0.0	0.0	0.0
240	92.3	79.7	2.0	-120.	-132.	12.	0.00380	0.00373	0.00007	0.0	0.0	0.0	0.0
241	143.1	130.5	7.0	-73.	-85.	12.	0.00407	0.00400	0.00007	0.0	0.0	0.0	0.0
242	244.8	232.2	4.0	21.	10.	12.	0.00460	0.00454	0.00007	0.0	0.0	0.0	0.0
243	219.6	207.0	4.0	-2.	-14.	12.	0.00447	0.00440	0.00007	0.0	0.0	0.0	0.0

## Appendix 1: Print of "pdprop.env" Simulation Control file

```
# This file contains the starting filenames, variables etc
# for the Crack Propagation programs. It should be edited by the
# user before each simulation run. It can also be generated from web
# page at: to be determined
#

#TYPE= plate_edge_flaw      #with or without weld using ACTIVATES:
#ACTIVATE_MmMb= 1 # Deactivate = 0
#ACTIVATE_MkmMkb= 0 # Note used in plate_edge_flaw
#ACTIVATE_fw= 0 # Note used in plate_edge_flaw
#
#                                #Other #TYPE= options:
#                                # plate_surface_flaw
#                                # plate_tru_flaw
#                                # plate_embedded_flaw
#                                # plate_long_surface_flaw
#                                #
#                                # pipe_inside_flaw
#                                # pipe_full_inside_flaw
#                                # pipe_full_outside_flaw
#                                #
#                                # rod_surface_flaw
#                                # rod_full_outside_flaw
```

## Results for filtExample\_2.5 : Crack Propagation Plate Edge Flaw

```
#
#                                     # These problem types are used to activate
#                                     # appropriate Fw, Mm, Mb, files etc.

# The factors described in this section may be ignored if not applicable to
# the particular problem type described above.
# (All dimensions in mm)
#B= 10.0    # plate (or pipe wall) thickness
#W= 70.0    # plate width
#ri= 00.    # Internal diameter if pipe problem. Ignored if not pipe
#azero= 1.5  # initial crack depth
#czero= 0.0  # initial 1/2 crack width at surface
#L= 00.     # Weld Feature width. Ignored if ACTIVATE_MkmMkb= 0 (above)

#HISTORYFILE= load1.txt    # historyFileName
#
# Adjustments to load file variables:
#
# Note that the MEANADD (below) is added AFTER the MAGFACTOR is applied.
#MAGFACTOR_m= 1.0    # Multiply factor on membrane load. Result should be MPa
#MAGFACTOR_b= 1.0    # Multiply factor on bending load term. Result should be MPa
#MEANADD_m= 0.0    # Mean shift in MPa added to membrane stress.
#MEANADD_b= 0.0    # Mean shift in MPa added to bending stress.

#MAXREPS= 1000000    # Max no. history repeats in simulation.
#
# One repetition or application of the load history is
# also called a "block" of cycles.
#
#
#MATERIAL= merged_a36_fitted.html    #File name of material fitted data
#
# This file is used to define the cyclic
# stress-strain curve, and the Neuber Product curve.
#
#DADN= table    # Can be "table" or "Paris"
#DADN_PARIS= 0.0 0.0 0.0 0.0 none    # Kth a m Kc units (ignored if #DADN= table )
#DADN_TABLE= a36+1015.dadn    # da/dN digitized da/dN curve for material,
#
# including the threshold, and KIC.
# If a threshold exists, put in a vertical line
# (with two identical X-axis points).
# If the threshold needs to be "turned off" then
# do NOT put in a vertical line at low da/dN.
# (Ignored when #DADN= PARIS )
#
#FAD Stuff:
#TensileFile= a36_Mattos_mono_engrSS_FLAT.txt    #enter "none" if no FAD
#PmEOL= 70.    #Set these so that Pm+Pb= 0.82*Syield for default.
#PbEOL= 100.
#Kmat= 1675.
#PinJoint= 0    # not used for plateEdgeFlaw.f
#
#BLOCKSKIP= 1.0 percent    # At the end of each block check if the previous
#
# two blocks of cycles had similar damage (crack
# extension) within this percentage. If TRUE then
# simply skip the simulation of the next block,
# but just add the expected damage. Continue by
# simulating the block after the skip.
# A value of 0.0 will disallow skipping blocks.
#SAVELEVEL= 0    #Amount of output saved to disk:
#
# 3=lots 2=medium 1=minimal
#
# 0= save #crk= data into binary direct access file only
#
# No #crk= data will be written into the text logfile.
#
# Use for large output files with lots of cycles.
```

## Appendix 2: Print of da/dn vs DeltaK Table in file filtExample\_2.5

Delta_K	da/dN						
0.1502160E+03	0.9620540E-07	0.2176716E+01	-0.7016800E+01	0.0000000E+00	0.0000000E+00	1	
0.1769830E+03	0.4562300E-06	0.2247931E+01	-0.6340816E+01	0.7121539E-01	0.6759844E+00	2	
0.2202350E+03	0.1160170E-05	0.2342886E+01	-0.5935478E+01	0.9495497E-01	0.4053378E+00	3	
0.2874840E+03	0.3224090E-05	0.2458614E+01	-0.5491593E+01	0.1157272E+00	0.4438853E+00	4	
0.4331670E+03	0.1069760E-04	0.2636655E+01	-0.4970714E+01	0.1780417E+00	0.5208793E+00	5	
0.7637410E+03	0.7556810E-04	0.2882946E+01	-0.4121662E+01	0.2462907E+00	0.8490520E+00	6	
0.1240590E+04	0.8520410E-03	0.3093628E+01	-0.3069540E+01	0.2106822E+00	0.1052122E+01	7	
0.1471680E+04	0.3307300E-02	0.3167813E+01	-0.2480526E+01	0.7418513E-01	0.5890131E+00	8	
0.1675690E+04	0.1074680E-01	0.3224194E+01	-0.1968721E+01	0.5638027E-01	0.5118057E+00	9	

## Appendix 3: Print of Stress-Strain-Init.Life file: "matfile"

#SAE Standard Fatigue Data File format

##

Pick one: #FDE\_plot #FDE\_fit ##

```
#
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#Foundation, Inc., 59 Temple Place - Suite 330, Boston, MA 02111-1307, USA
#Try also their web site: http://www.gnu.org/copyleft/gpl.html
#
# NOTE: Fitted Data !!
# A36 Steel Merged Data Sets from Refs. 1 and 2:
# Ref.1: P.Dindinger report to Fat.Des.+Eval. Comm. Apr.2012
# Ref.2: G.A.Miller and H.S.Reemsnyder, "Strain-Cycle Fatigue of Sheet and
# Plate Steels I: Test Method Development and Data Presentation,"
# SAE Paper 830175, Detroit MI, Feb28-Mar.4, 1983
#
# NOTE that original test data ends at 2Nf = 1.3million.
#
#FileType= strain_life
#DataType= fitted
#TIMEcol= 0
#NAME= ASTM-A36
#NAME= Structural
#NAME= Steel
#Stress_units= ksi
#Strain_units= strain
#Sy= 38.4 0.2pc offset, 265 mpa
#Su= 69. ksi from Miller/Reemsnyder = 475 mpa
#eu= 0 #strain at Su not reported
```

## Results for filtExample\_2.5 : Crack Propagation Plate Edge Flaw

```
#E= 29528. ksi = 203600 mpa
#FractureStrain= 0 not reported
#FractureStress= 0. not reported
#monotonic_K= 0 not reported
#monotonic_n= 0 not reported
#BHN= 138.
#%RA= 0. % not reported
#
#saedigcurve_v2.2.f starts.
# NOTE!! The Following Points are FITTED DATA:#NOTE!! Fitted Stress computed using Experm.
# Total Strain 2Nf Stress Mean Plastic Strain Initial
# Amp Amp Stress Amp Elastic Mod.
0.88485 1 115.3 0. 0.88095 29528. #Fitted_point
0.00914 5000 52.1 0. 0.00737 29528. #Fitted_point
0.00665 10000 48.8 0. 0.00499 29528. #Fitted_point
0.00493 20000 45.7 0. 0.00338 29528. #Fitted_point
0.00344 50000 42.0 0. 0.00202 29528. #Fitted_point
0.00270 100000 39.3 0. 0.00136 29528. #Fitted_point
0.00217 200000 36.8 0. 0.00092 29528. #Fitted_point
0.00169 500000 33.8 0. 0.00055 29528. #Fitted_point
0.00144 1000000 31.6 0. 0.00037 29528. #Fitted_point
#Original test data ends at 2Nf = 1.3million.
#Points below are extrapolation:
0.00125 2000000 29.6 0. 0.00025 29528. #Fitted_point
0.00106 5000000 27.1 0. 0.00014 29528. #Fitted_point
#
#
```