# AISI 1541 Steel Iteration #1

# Fatigue Behavior, Monotonic Properties and Microstructural Data

Prepared by:

A. Varvani-Farahani, A. Dabayeh and T.H. Topper

Department of Civil Engineering University of Waterloo Waterloo, Ontario Canada

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The AISI Bar Steel Applications Group

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### **SUMMARY**

The required chemical analysis, microstructural data, mechanical properties, cyclic stress-strain data and strain-controlled fatigue data for AISI 1541 normalized steel have been obtained. The material was provided by the American Iron and Steel Institute (AISI) in the form of metal bars. These bars were machined into smooth axial fatigue specimens. A monotonic tensile test was performed to measure yield strength, tensile strength and reduction of area. Twenty three specimens were fatigue tested in laboratory air at room temperature to establish a strain-life curve.

### INTRODUCTION

This report presents the results of tensile and fatigue tests performed on a group of 24 normalized 1541 steel samples. The material was provided by the American Iron and Steel Institute.

The objectives of this investigation were to obtain a chemical analysis, and the microstructural data, mechanical properties, cyclic stress-strain data and 18 strain-life tests requested by the AISI bar group.

### EXPERIMENTAL PROCEDURE

### **Specimen Preparation**

The material for the study was received in the form of bars. Smooth cylindrical fatigue specimens, shown in Figure 1, were machined from the metal bars. The gauge sections of the fatigue specimens were mechanically polished in the loading direction using 240, 400, 500, and 600 emery paper. After polishing, a thin band of M-coat D acrylic coating was applied along the central gauge section. The purpose of the M-coat D application was to prevent scratching of the smooth surface by the knife edges of the strain extensometer, thus reducing the incidence of knife-edge failures. In total, 18 fatigue data points were generated at 10% load drop and 5 extra data points were generated at 50 % load drop, to check the difference in fatigue life between 10% and 50% load drop criteria.

### **Test Equipment and Procedure**

A monotonic tension test was performed to determine the yield strength, the tensile strength, the percent of elongation and the percent reduction of area. Hardness tests were performed on the surface of three fatigue specimens using a "Rockwell B" scale. The hardness measurements were repeated three times for each specimen and the average value was recorded.

All fatigue tests were carried out in a laboratory environment at approximately 25 °C using an MTS servo-controlled closed loop electrohydraulic testing machine. A process control computer, controlled by FLEX software [1] was used to output constant strain and stress amplitudes in the form of a sinusoidal wave.

Axial, constant amplitude, fully reversed (R=-1) strain-controlled fatigue tests were performed on smooth specimens. The stress-strain limits for a given cycle of each specimen were recorded at logarithmic intervals throughout the test via a peak reading oscilloscope. Failure of a specimen was defined as a 10 percent drop in tensile peak load from the peak load observed at one half the expected specimen life. Five extra data points, one at each strain level, were generated at 50 percent load drop. For fatigue lives greater than 100,000 reversals, the specimens were tested in stress-control once the stress-strain loops had stabilized. For the stress-controlled tests, failure was defined as the separation of the smooth specimen into two pieces. For strain-controlled tests the loading frequency varied from 0.03 Hz to 5 Hz while in stress-controlled tests the frequency used was up to 30 Hz.

The first reversal of each fatigue test was recorded on an x-y plotter, allowing the elastic modulus (E) and the monotonic yield strength to be determined.

### RESULTS

### A) Microstructural Data

Figure 2 presents the ferritic-pearilitic microstructure of AISI 1541 normalized steel. The ASTM ferritic grain size number in the longitudinal (extrusion) direction and in the transverse direction is 10 according to ASTM E112. Type D inclusions with a rate of  $2\frac{1}{2}$  were obtained based on the severity level number according to ASTM E45 method A. Inclusions of types A, B, and C were not observed. Figure 3 presents the observed inclusions of AISI 1541 normalized steel. The inclusion area was measured using a JAVA image analysis system. The volume fraction of pearlitic structure based on ASTM E562-95 and using the JAVA image analysis system was 71%. The chemical composition of AISI 1541 normalized steel was provided by SCI-Lab materials testing inc., 25 McIntyre place, unit 2, Kitchener, Ontario, N2R 1H1, and is shown in table 1.

### B) Strain-Life Data

The fatigue test data for AISI 1541 normalized steel obtained in this investigation are given in table 2. The stress amplitude corresponding to each strain-amplitude was calculated from the peak load amplitude at one half the expected specimen life. Data for a 10% and a 50% load drop fell within the same scatter band.

A fatigue strain-life curve for the AISI 1541 normalized steel is shown in Figure 4, and may is described by the following equation:

$$\frac{\Delta \varepsilon}{2} = \frac{\sigma_f'}{F} (2N_f)^b + \varepsilon_f' (2N_f)^c$$

where

 $\frac{\Delta \varepsilon}{2}$  = True total strain amplitude

 $2N_f$  = Number of reversals to failure

 $\sigma'_{\mathbf{f}}$  = Fatigue strength coefficient

**b** = Fatigue strength exponent

 $\mathbf{E}'\mathbf{f}$  = Fatigue ductility coefficient

**c** = Fatigue ductility exponent

Where  $\sigma'_{\mathbf{f}} = 1237$  MPa,  $\mathbf{b} = -0.1075$ ,  $\varepsilon'_{\mathbf{f}} = 0.369$  and  $\mathbf{c} = -0.512$ . These values of the strain-life parameters were determined from fatigue testing over the range:  $0.002 < \frac{\Delta \varepsilon}{2} < 0.01$ .

### C) Cyclic Stress-Strain Curves

Stabilized and half life stress data obtained from strain-life fatigue tests were used to obtain the companion cyclic stress-strain curve shown in figure 5. The true cyclic stress-strain curve is described by the following equation:

$$\varepsilon = \frac{\sigma}{F} + \left(\frac{\sigma}{K'}\right)^{\frac{1}{n'}}$$

where

 $\varepsilon$  = True total strain amplitude

 $\sigma$  = Cyclically stable true stress amplitude

**K'** = Cyclic strength coefficient

n' = Cyclic strain hardening exponent

Where K' = 1399.80 MPa and n' = 0.1934.

### D) Mechanical Properties

The engineering monotonic stress-strain curve is given in figure 6. The monotonic and cyclic properties are included in Appendix 1. The Hardness of the AISI 1541 normalized steel taken as the average of three randomly chosen fatigue specimens is given in Appendix 1. The individual hardness measurements are also given in Table 2. The true monotonic and true cyclic stress-strain curves plotted together are given in figure 7.

### REFERENCES

- [1] Pompetzki, M.A., Saper, R.A., and Topper, T.H., "Software for High Frequency Control of Variable Amplitude Fatigue Tests," Canadian Metallurgical Quarterly, Vol. 25, No. 2, pp. 181-194, 198.
- [2] J. A. Bannantine, J. J. Comer, and J. L. Handrock (1990), In: Fundamentals of Metal Fatigue Analysis, Prentice Hall, London.

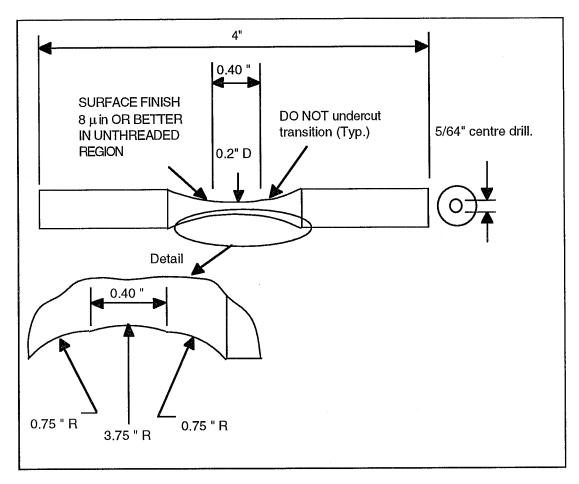


Fig. 1 Smooth cylindrical fatigue specimen

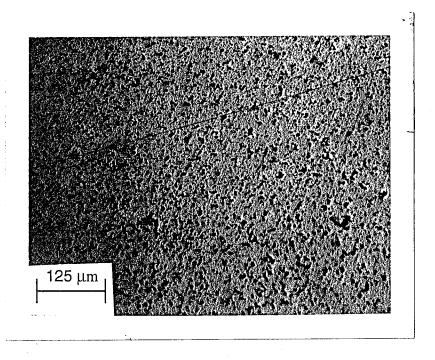


Fig. 3 Inclusions photomicrograph of AISI 1541 normalized steel (X200)

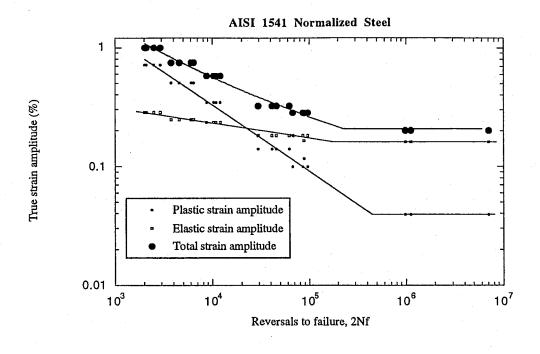


Fig. 4 Constant amplitude fully reversed strain-life curve for AISI 1541 normalized steel

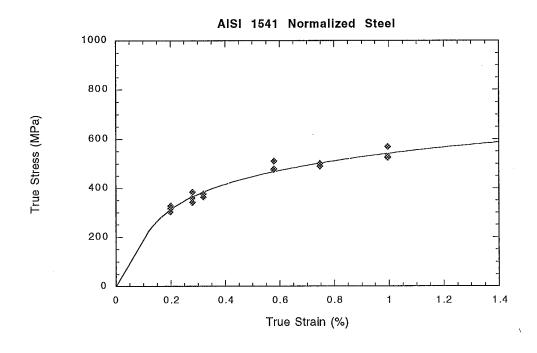


Fig. 5 Cyclic stress-strain curve for AISI 1541 normalized steel

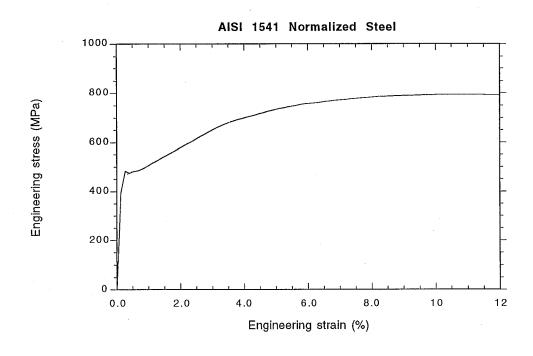


Fig. 6 Monotonic stress-strain curve for AISI 1541 normalized steel

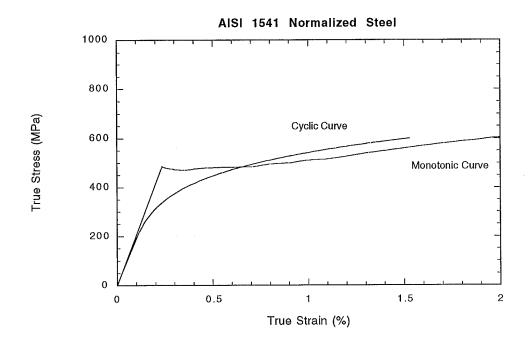


Fig. 7 Monotonic and Cyclic stress-strain curves for AISI 1541 normalized steel

Table 1 Chemical composition of AISI 1541 Normalized Steel

Carbon, C	0.41%
Manganese, Mn	1.37%
Phosphorous, P	0.009%
Sulfur, S	0.098%
Silicon, Si	0.17%
Copper, Cu	0.1%
Nickel, Ni	0.05%
Chromium, Cr	0.07%
Molybdenum, Mo	0.01%
Vanadium, Va	0.002%
Calcium, Ca	0.001%
Boron, Bo	0.0029%
Aluminum, Al	<0.001%
Titanium, Ti	0.001%
Oxygen, O	0.0029%
Columbium, Cb	0.002%

Table 2 Tensile and Fatigue Test Data for AISI 1541 Normalized Steel

	Hardness	(HRB)	92			91	92				92					91.5				92				91	
ENG.	Load Drop	(%)	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	10	50	50	50	50	50
	MONOTONIC	Modulus(GPa)	190	205	189	205	189	195	200	202	192	202	200	198	193	190	198	200	191	195	200	210	191	961	199
	Fatigue Life	(Reversals, 2Nf)	3756	29714	8740	88224	2922	85412	7075200	99029	45986	12086	10898	6383	6030	2006	2500	67486	1119300	988810	2100	41000	10314	95964	4546
TRUE	Elastic Strain	Amplitude(%)	0.254	0.185	0.240	0.172	0.291	0.193	0.156	0.189	0.189	0.241	0.238	0.252	0.251	0.261	0.265	0.181	0.165	0.152	0.194	0.224	0.234	0.193	0.291
TRUE	Plastic Strain	Amplitude(%)	0.493	0.134	0.338	0.108	0.705	0.086	0.040	0.130	0.129	0.337	0.340	0.495	0.496	0.735	0.731	0.097	0.034	0.047	0.801	0.095	0.344	0.086	0.457
TRUE	Stress Amplitude	(MPa)	506.12	367.59	477.52	339.79	580.99	383.25	315.83	375.50	375.50	479.51	473.56	502.15	500.16	521.30	529.26	359.54	327.67	302.01	387.99	444.67	465.64	383.25	579.55
TRUE	Total Strain	Amplitude(%)	0.747	0.319	0.578	0.279	0.995	0.279	0.199	0.319	0.319	0.578	0.578	0.747	0.747	0.995	0.995	0.279	0.199	0.199	0.995	0.319	0.578	0.279	0.747
	Spec		5	26	7	27	78	19	14	17	29	13	23	3	18	∞	7	20	6	21	12	16	11	24	15

# Appendix 1

# **Monotonic Properties for AISI 1541 Normalized Steel**

Average Elastic Modulus, E	=	197	GPa
Upper Yield Strength	=	484.17	MPa
Lower Yield Strength	=	471.20	
Ultimate tensile Strength	<u>.</u>	783.21	MPa
% Elongation	=	38.66	%
% Reduction of Area	=	55.11	%
True fracture strain, $Ln(A_i/A_f)$	=	80	%
True fracture stress, $\sigma_f = \frac{P_f}{A_f}$		1468 M	
Bridgman correction, $\sigma_f = \frac{P_f}{A_f} / \left(1 + \frac{4R}{D_f}\right) Ln$	$\left(1+\frac{D_f}{4R}\right)$	=1206.	8 MPa
Monotonic strength coefficient, K	= "	1165 M	1Pa
Monotonic strain hardening exponent, n	=	0.143	
Hardness, Rockwell B (HRB)	=	91	
Hardness, Brinell	= .	180	

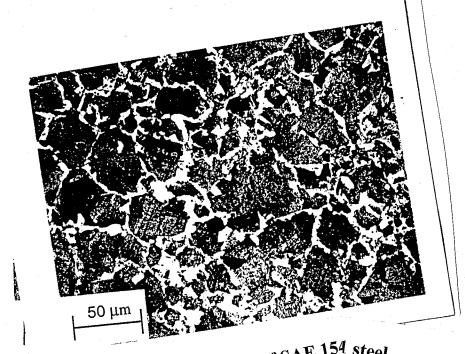
# **Cyclic Properties for AISI 1541 Normalized Steel**

Cyclic Yield Strength, $(0.2\% \text{ offset}) = K'(0.2\% \text{ offset})$	$(0.002)^{n'}$	=421 MPa
Cyclic strength coefficient, K'	=	1399.8 MPa
Cyclic strain hardening exponent, n'	_	0.1934
Fatigue Strength Coefficient, $\sigma'_f$	=	1237 MPa
Fatigue Strength Exponent, b	=	-0.1075
Fatigue Ductility Coefficient, ε'f	= '	0.369
Fatigue Ductility Exponent, c	. =	-0.512

Load at fracture.

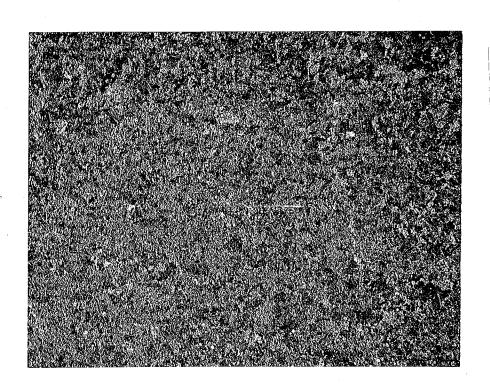
P<sub>f</sub>:
A<sub>i</sub> and A<sub>f</sub>:
R:
D<sub>f</sub>

Specimen cross-section area before and after fracture. Specimen neck radius.
Specimen diameter at fracture.

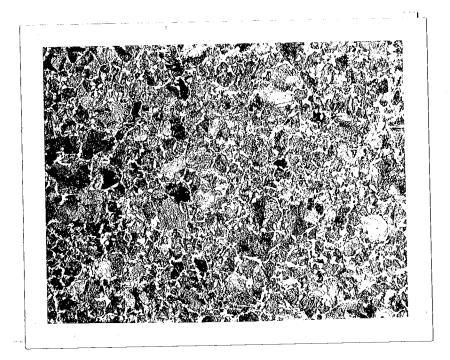


ITER 1: Photomicrograph of SAE 154 steel,
500X Mag.
Normalized to Rb-91.

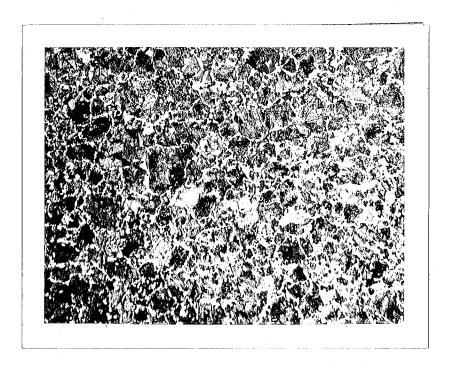
(a) Longitudinal direction



ITER 1: Photomicrograph of SAE 1541 steel, Normalized to Rb-91. 100X Mag.



(a) Longitudinal direction (2000)



(b) Transverse direction

Fig. 2 photomicrographs of AISI 1541 normalized steel (X500)

# AISI Bar Application Group Fatigue Project Test Matrix (6-9-98)

Bar or Part	HR Bar HR Bar	Hr Bar Axles Axles	Hr Bar Stab Bar Stab Bar Stab Bar Stab Bar	Hr Bar Hr Bar	Hr Bar Hr Bar	Hr Bar Hr Bar Hr Bar	Hr Bar Hr Bar Hr Bar	Hr Bar Hr Bar
School	Waterloo Waterloo	Waterloo Waterloo Waterloo	Waterloo Waterloo Waterloo Waterloo	Toledo Toledo	Toledo Toledo	Toledo Toledo Toledo	Toledo Toledo Toledo	Toledo Toledo
Microstrücture	Ferrite/Pearlite Ferrite/Pearlite	Ferrite/Pearlite Martensite Martensite	Pearlite Martensite/Bainite Martensite Martensite/Bainite Bainite	Ferrite/Pearlite Martensite	Ferrite/Pearlite Martensite	Ferrite/Pearlite Martensite Ferrite/Pearlite	Ferrite/Pearlite Ferrite/Pearlite Martensite	Ferrite/Pearlite Ferrite/Pearlite
Fabricated Hdns	Rb 85 min.	Rc 35 max. Rc 58 min.	BHN 341-444 BHN 341-444 BHN 302-363 BHN 3.0-3.5	BHN 229-269	BHN 229-269	BHN 229-269 TBD	Rb 85 min. Rc 20-30	RC 25-30
Part Fabrication	Normalize 1650F Cold Size/Form	Normalize 1650F Hot Forge, Cold Extrude (Core) Induction Surface Hardened (Case)	Normalize 1650F Hot Form + Acc. Cool Hot Form + Q&T Hot Form + Acc. Cool Hot Form + Austemper	Normalize 1650F Reheat, Q&T	Normalize 1650F Reheat, Q&T	Normalize 1650F Reheat, Q&T Normalize @ High Temp (1750F)	Normalize 1650F Cold Size/Form Reheat, Q&T, (Temper @ 930F)	Normalized 1650F Reheat, 2250F, Deform @ 2000F, FAC
Status	Complete 2-4-98 Complete 2-4-98	Complete 7-30-97 Complete 6-9-98 Complete 6-9-98	Complete 6-9-98 Complete 6-9-98 Complete 2-4-98 Complete 2-4-98 Complete 6-9-98	Complete 6-9-98 Complete 6-9-98	Complete 6-9/98 Complete 6-9-98	@ Toledo @ Toledo @ Toledo	Complete 10-9-97 Complete 6-9-98 @Toledo-Re-H.T.	@Toledo @Toledo
Supplier	RES	Stelco Chrysler Chrysler	NS Chrysler Chrysler GM/AMM GM/Mather	RES RES	S S S	8 8 8 8 8 8	Stelco Stelco Stelco	Inland Inland
<u>Steel</u>	1541 1541	1050M 1050M 1050M	1090 1090M 1090 1090M	1141(AIFG) 1141(AIFG)	1141(NbFG) 1141(NbFG)	1141(VFG) 1141(VFG) 1141(VFG)	1038 1038 1038	10V45 10V45
Iter. No.	1 2	ω4ω	ο / α * <sub>0</sub> 0	112	6 4	15 16 17	18 19 20	21

\*These iterations will be supplied by GM, and funded separately from the rest of the program.